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PREPWATE—An Interactive Preprocessing Computer Code to the Weight Analysis of Turbine Engines (WATE) Computer Code

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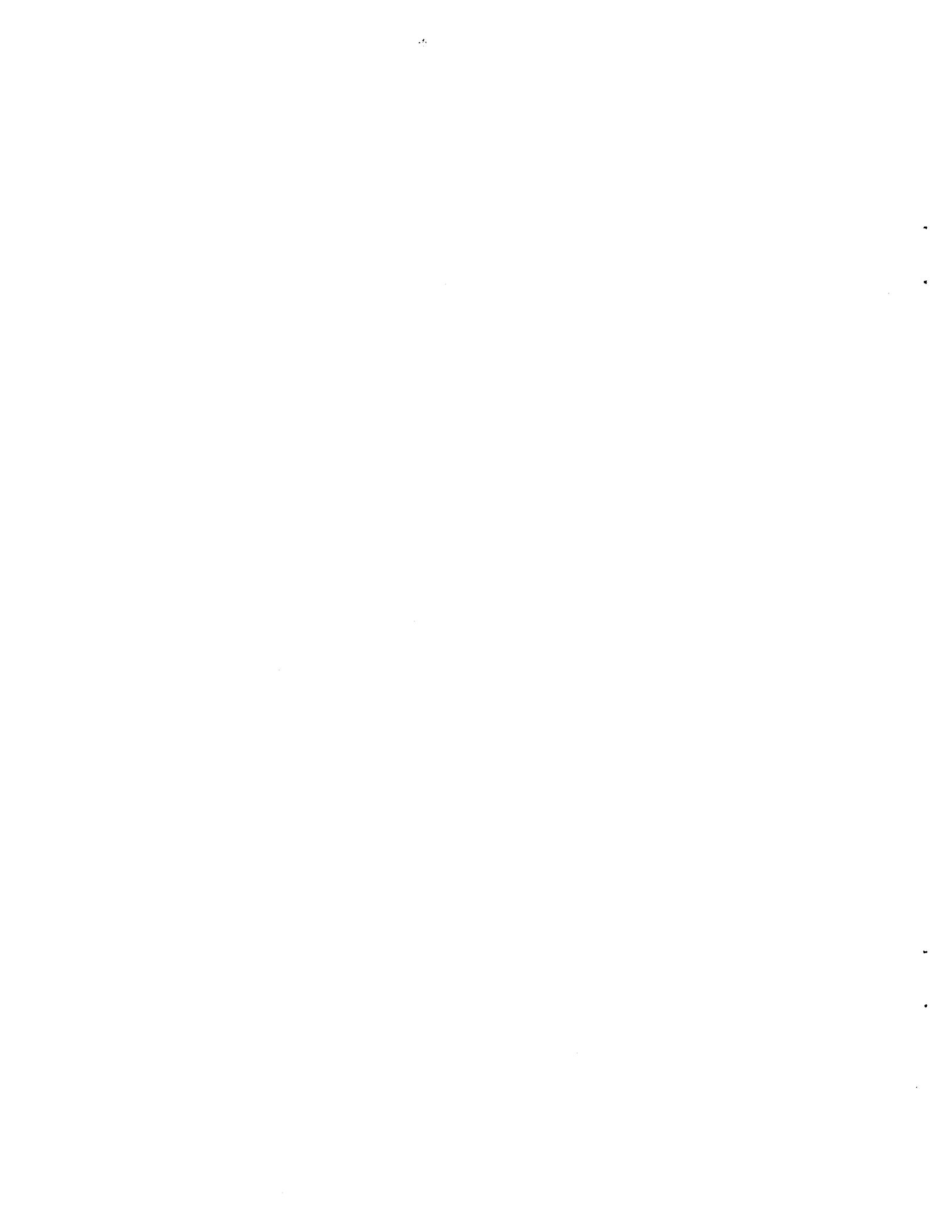
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UTL: PREWATE: An interactive preprocessing computer code to the Weight
Analysis of Turbine Engines (WATE) computer code

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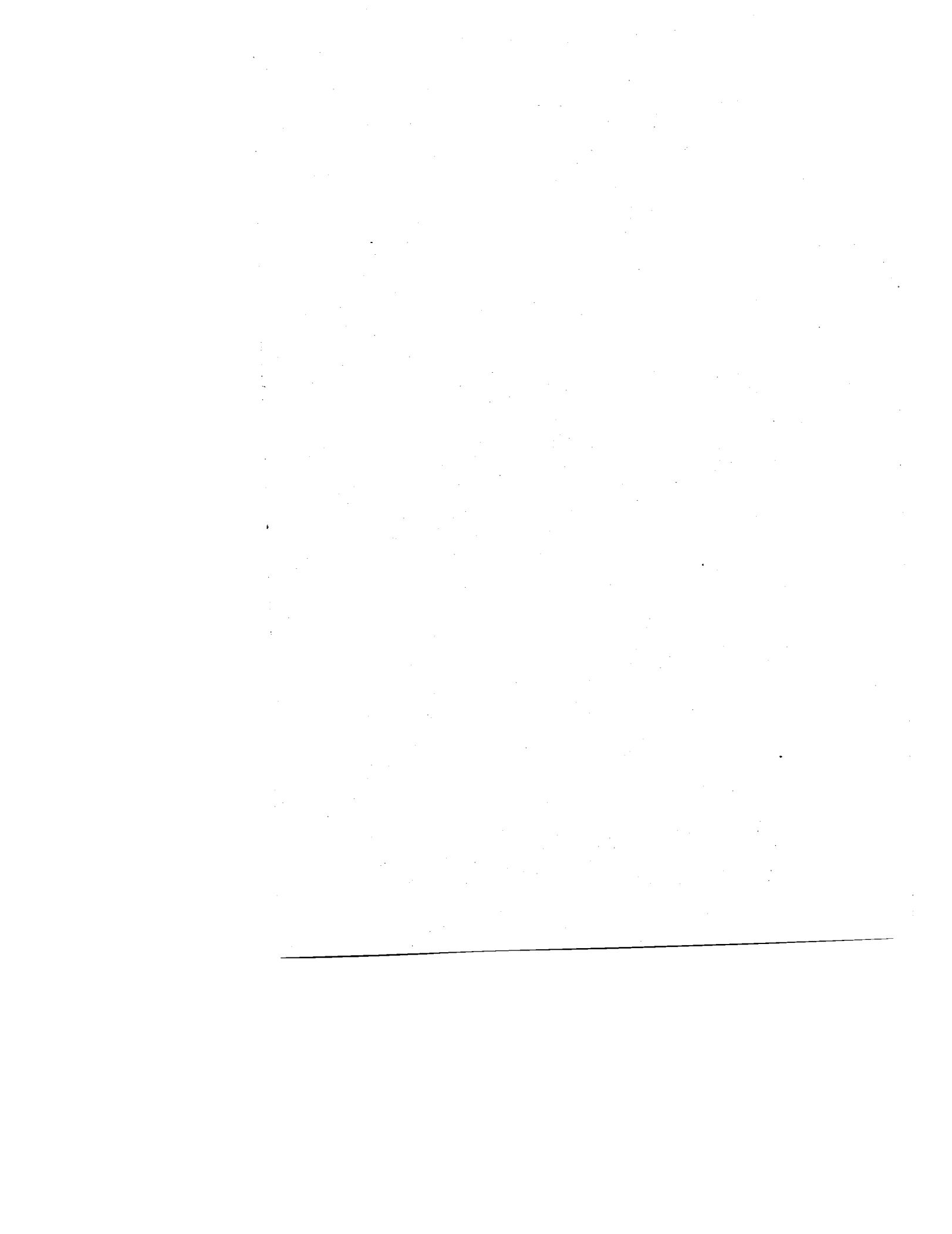
MAVS: /*COMPUTER PROGRAMS/*PREDICTIONS/*TURBINE ENGINES/*TURBOFAN ENGINES/*
WEIGHT (MASS)

MINS: /* COMPRESSORS/* DESIGN ANALYSIS/* THERMODYNAMIC PROPERTIES/* THRUST

ABA: Author

ABS: The Weight Analysis of Turbine Engines (WATE) computer code was developed
by Boeing under contract to NASA Lewis. It was designed to function as an
adjunct to the Navy/NASA Engine Program (NNEP). NNEP calculates the design
and off-design thrust and stc performance of User defined engine cycles.
The thermodynamic parameters throughout the engine as generated by NNEP
are then combined with input parameters defining the component
characteristics in WATE to calculate the bare engine weight of this User
defined engine. Preprocessor programs for NNEP were previously developed
to simplify the task of creating input datasets. This report describes a
similar preprocessor for the WATE code.

ENTER:



PREPWATE - AN INTERACTIVE PREPROCESSING
COMPUTER CODE TO THE WEIGHT ANALYSIS OF TURBINE
ENGINES (WATE) COMPUTER CODE

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SUMMARY

The Weight Analysis of Turbine Engines (WATE) computer code was developed by Boeing under contract to NASA Lewis (ref. 1). It was designed to function as an adjunct to the Navy/NASA Engine Program (NNEP) (ref. 2). NNEP calculates the design and off-design thrust and sfc performance of User defined engine cycles. The thermodynamic parameters throughout the engine as generated by NNEP are then combined with input parameters defining the component characteristics in WATE to calculate the bare engine weight of this User defined engine.

Preprocessor programs for NNEP were previously developed (ref. 3) to simplify the task of creating input datasets. This report describes a similar preprocessor for the WATE code.

INTRODUCTION

The NASA Lewis Research Center in conjunction with the Naval Air Development Center jointly developed a computer code (NNEP) for determining the thermodynamic performance for arbitrary turbine engines, i.e., the code can assemble arbitrary combinations of specified types of components (such as ducts, compressors, turbines, etc.) through the use of input variables rather than having to build a computer code exclusively for each engine configuration (ref. 2). In order to simplify the task of creating datasets to run on NNEP for new or occasional users, a pair of preprocessor programs KONFIG and REKONFIG were created (ref. 3). Favorable comments were received from users of the preprocessor programs. They also expressed the desire for a similar preprocessor for the Weight Analysis of Turbine Engines code (WATE) (ref. 1).

WATE was developed by Boeing under contract to NASA Lewis. It is capable of calculating the component weights and flowpaths of most engines that a user of NNEP would wish to study. It was designed to be an adjunct to NNEP and receives thermodynamic data from NNEP. Mechanical design variables are then set for each of the engine components and the weight and dimensions then determined.

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As opposed to NNEP, which treats all of a particular type of component (e.g., compressors) thermodynamically the same, WATE distinguishes them by type (e.g., fan, low pressure, high pressure). Different modules then calculate the weight by sub-type. The WATE code requires two main arrays of data, IWMEC and DESVAL. IWMEC is a two dimension array where the mechanical design inputs such as component type, the existence of frames, stators, and other information are stored. DESVAL is also a two dimensional array containing the aerodynamic and design inputs such as Mach number in, Mach number out, blade aspect ratio, etc. Default values of DESVAL for each component sub-type are built into PREPWATE. When the user runs PREPWATE, he will be shown the defaulted values as well as the recommended range for each variable. He may easily change one or more of the DESVALs for each component in the engine.

The rest of this report will show the operation of PREPWATE by a sample case.

OPERATION OF PREPWATE

PREPWATE consists of the main and one subroutine. The main routine prompts the User for all of the inputs required and prepares the output data-set. The subroutine (SUBROUTINE REDOEM) is called in order to allow the User to change one or more of the DESVALs from the defaulted values.

All reads are from UNIT 30 and prompts for input on UNIT 35. The final card images for the inputs to the WATE code are written on UNIT 6 and a data-set should be dataeffed to this unit to save the images.

SAMPLE CASE

A high bypass ratio separate flow engine will be used to illustrate the use of the code. The fan will have booster stages attached to it to raise the pressure of the air that will enter the core. These stages are driven by the low pressure turbine and the outer dimension of both turbines will be limited to the outer dimension of these booster stages. An NNEP thermodynamic schematic of the engine is shown in figure 1. As indicated, NNEP does not distinguish by subcomponent type, e.g., all compressor thermodynamic calculations are done identically.

The WATE code, on the other hand, requires that the User identify the type of subcomponent, e.g., the three compressors are now a fan, a low pressure compressor, and a high pressure compressor. The mechanical schematic for the WATE code for the NNEP engine in figure 1 is shown in figure 2. The component numbers are retained so there is a one to one correspondence with the thermodynamic model. Note, however, the inlet does not appear in figure 2. Inlet weight cannot be calculated by WATE. At NASA Lewis, this weight is calculated by the INSTAL computer code (ref. 4) which was also created as an adjunct to NNEP.

The User is now ready to run PREPWATE. A terminal session for creating the WATE input data for this engine is shown in Appendix A.

As can be seen from Appendix A, the user first defines a dataset on Unit 6 into which the card images for the WATE code will be written. He then identifies the library containing the Main Routine and Subroutine and then calls the main routine. (A procedure definition containing these instructions has been prestored in the LeRC 370/3033 as PREPWATE).

The User is then prompted for the required inputs. The first inputs specify what plots and whether English or SI units are desired. The second inputs are whether or not scaled engines are desired (IWT = 4) or not (IWT = 2) and for the amount of output desired (IOUTCD = 4 gives total weight, length and maximum radius, plus component values, plus stage by stage values).

Next, the disk weight calculation method to be used is identified (B = Boeing, G = Garrett). The User is then asked to identify the component numbers of all the components contributing to the maximum length of the engine (exclusive of inlets and water injectors). All the remaining flow components in the engine including the shafts, but excluding inlets and water injectors are then identified.

The User will then be prompted to enter the NNEP generic type of each component (COMP, TURB, NOZZ, etc.) and then for detailed information about each component (type of compressor, are there stators?, frames, etc.). After this information has been entered, the code will then print out the defaulted values for each DESVAL for each subcomponent as well as the recommended range (where appropriate). The User can change any of the defaulted values of DESVAL by entering the letter code for the variable followed by the new value which MUST include a decimal point. Any or all of the variables may be changed in this manner. By entering an "R", the User may review the latest values. When satisfied, the User enters a "T" to terminate this component and proceed to the next. When all the components are done, the program will terminate.

The card images generated from PREPWATE are shown in Appendix B. The extra blanks generated for DESVAL have been eliminated. This dataset must be incorporated into a NNEP dataset in order to run the NNEP/WATE combination. This dataset is also shown in Appendix B. A brief explanation of this dataset is as follows:

The first card image is the NNEP Title Card;

Next, the first block of Namelist D data. The defaulted values for all variables are being used.

The second block of Namelist D data for NNEP is then displayed. This represents the configuration and design spec data for Mode 1. (There is only one mode in this engine). There is an IWT = 4 card which tells the code that weight should be calculated after this NNEP point has been run. This triggers the code so that the WATE code is called.

The WATE code calls for the Namelist W data and the card images created by PREPWATE are entered here. The NNEP/WATE code is now ready to run. The answers to the sample case are shown in Appendix C. The interactive graphics capability of the code (ref. 5) was used to generate a graphics plot of the engine (PLOT = T in & W). This plot has been reproduced in figure 3. Note that the LPC (component 6) overlaps the duct (component 4). Changes to the DESVALS of components 2, 4 and/or 6 can be implemented to generate a better flowpath, either by going back to the original dataset or using the interactive method of reference 5 at the graphics scope.

The PREPWATE computer code listings are shown in Appendix D.

CONCLUSIONS

PREPWATE has been created to help Users of the WATE code to prepare input datasets for the code. It is tutorial in nature and hopefully will prove to be as useful as the preprocessors KONFIG and REKONFIG for the NNEP computer code. The code is available from the author in card form (600 cards) or on a User supplied tape.

55

```
DOEF FT06F001, VS, WATEDATA
DOEF xx, VP, LIB, PREPWATE, OPTION & JOBLIB

GO:PREPWATE
ENTER VALUES FOR IPLT(PRINTER PLOT), PLOT(GRAPHICS PLOT), ISII(SI INPUT), ISIO(SI OUTPUT) -- T OR F
T T F F
ENTER VALUES FOR IWT (2=NO AIRFLOW SCALING,4=WITH AIRFLOW SCALING) AND
IOUTCD (0=WT,L, & MAX R,1=ADD COMPONENT WTS,2=ADD STAGE BY STAGE OUTPUT
6 2
DO YOU WANT BOEING OR GARRETT METHOD FOR DISK WEIGHT CALCULATIONS?
ENTER B FOR BOEING OR G FOR GARRETT
B
ENTER COMPONENT NUMBERS OF ALL COMPONENTS (EXCLUSIVE OF INLETS AND WINJS) THAT CONTRIBUTE TO MAX LENGTH OF ENGINE (RT. A
DJ.)
02 03 06 07 08 17 09 10 11 12 13 14 -----
ENTER COMPONENT NUMBERS OF ALL REMAINING COMPONENTS (EXCLUSIVE OF INLETS,WINJS,LOADS,CTLRS,OPTVS, & LIMVS)
04 05 15 16 -----
YOU WILL NOW BE PROMPTED FOR THE GENERIC (MHEP) TYPE FOR EACH COMPONENT (COMP,TURB, ETC.)
COMPONENT 2 IS A (A4)
COMPONENT 3 IS A (A4)
SPLT
COMPONENT 6 IS A (A4)
COMPONENT 7 IS A (A4)
DUCT
COMPONENT 8 IS A (A4)
COMPONENT 17 IS A (A4)
DUCT
COMPONENT 9 IS A (A4)
DUCT
COMPONENT 10 IS A (A4)
TURB
COMPONENT 11 IS A (A4)
DUCT
COMPONENT 12 IS A (A4)
TURB
COMPONENT 13 IS A (A4)
DUCT
COMPONENT 14 IS A (A4)
NOZZ
COMPONENT 4 IS A (A4)
```

DUCT
COMPONENT 5 IS A (A4)

NOZZ
COMPONENT 15 IS A (A4)

SHFT
COMPONENT 16 IS A (A4)

SHFT
FOR EACH NHEP COMPONENT YOU WILL BE PROMPTED FOR THE NATE SUBCOMPONENT
COMPONENT 2 IS A COMPRESSOR, OPTIONS ARE (ENTER CORRECT LETTER)

- A- TYPICAL FAN
- B- OUTER PORTION OF NON-ROTATING SPLITTER FAN
- C- INNER PORTION OF NON-ROTATING SPLITTER FAN
- D- OUTER PORTION OF ROTATING SPLITTER FAN
- E- INNER PORTION OF ROTATING SPLITTER FAN
- F- LOW PRESSURE COMPRESSOR
- G- HIGH PRESSURE COMPRESSOR

A
INDICATE WHETHER THERE ARE STATORS OR IF THIS IS A CENTRIFUGAL COMPRESSOR

- A- CALCULATE STATOR WEIGHT
- B- NO STATOR WEIGHT
- C- CENTRIFUGAL COMPRESSOR

B
INDICATE WHETHER THERE IS A FRONT FRAME IN THE COMPRESSOR

- A- NO FRAME
- B- SNGL BEARING FRAME FOR TFS AND TJS WITHOUT POWER TAKEOFF (PTO)
- C- SINGLE BEARING FRAME WITH PTO
- D- TWO BEARING FRAME WHICH EXTENDS OUTWARD TO THE FAN OUTER CASE AND HOLDS TWO BEARINGS WITH PTO

A
INDICATE WHETHER THERE IS A REAR FRAME IN THE COMPRESSOR

- A- NO FRAME
- B- SNGL BEARING FRAME FOR TFS AND TJS WITHOUT POWER TAKEOFF (PTO)
- C- SINGLE BEARING FRAME WITH PTO
- D- TWO BEARING FRAME WHICH EXTENDS OUTWARD TO THE FAN OUTER CASE AND HOLDS TWO BEARINGS WITH PTO

D
GEAR BOX INDICATOR, 0= NO GEAR BOX, N=SHAFT NUMBER FOR GEAR BOX

0
YOU MAY SPECIFY THE NUMBER OF STAGES-OVERRIDES MAX PR/STG., OTHERWISE ENTER 0

FAH/COMPRESSOR	CODE	VALUE	DESCRIPTION	LOW	VALUE-HIGH	VALUE
	A	0.5500	FACE INLET MACH NUMBER	0.5000	0.6000	
	B	1.70000	MAX 1ST STAGE PRATIO	1.50000	1.80000	
	C	0.450	COMPRESSOR FACE HUB TO TIP RATIO	0.400	0.500	
	D	1.5000	BLADE SOLIDITY (CORD/SPACING)	1.0000	1.5000	
	E	4.0000	BLADE ASPECT RATIO-1ST STAGE	3.0000	5.0000	
	F	3.0000	BLADE ASPECT RATIO-LAST STAGE	2.0000	3.0000	
	G	0.4500	COMPRESSOR EXIT MACH NUMBER	0.4500	0.5500	
	H	0.0000	MAX. COMP. INLET T- 0=CALC'D	0.0000	0.0000	
	I	0.0000	MAX. COMP. EXIT T- 0=CALC'D	0.0000	0.0000	
	J	1.0000	MAX. SPEED RATIO RPMMAX/RPM0	1.0000	1.0000	
	K	0.0000	BLADE MATERIAL DENSITY 0=T-SET	0.0000	0.0000	
	L	2.	1=CONST.HUB,2=MEAN,3=TIP DESH.	0.	0.	
	M	1.0000	RPM SCALER TO MATCH KNOWN RPM	1.0000	1.0000	
	N	0.0000	TEMP FOR MATERIAL CHANGE	0.0000	0.0000	
	O	0.0000	WEIGHT SCALER 0=AS CALCULATED	0.0000	0.0000	

P 1.8000 STATOR BLADE TAPER RATIO 0.0000 0.0000
S 0.0550 BLADE VOLUME RATIO 0.0000 0.0000

ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER
Q QUIT PROCESSING ENTIRE ENGINE
R REVIEW UPDATED VALUES
T TERMINATE - GO ON TO NEXT COMPONENT

B 1.8
C .37
E 4.5
F 4.5
G .5
R

FAN/COMPRESSOR

CODE	VALUE	DESCRIPTION	LOW VALUE	HIGH VALUE
A	0.5500	FACE INLET MACH NUMBER	0.5000	0.6000
B	1.80000	MAX 1ST STAGE PRATIO	1.50000	1.80000
C	0.370	COMPRESSOR FACE HUB TO TIP RATIO	0.400	0.500
D	1.5000	BLADE SOLIDITY (CORD/SPACING)	1.0000	1.5000
E	4.5000	BLADE ASPECT RATIO-1ST STAGE	3.0000	5.0000
F	4.5000	BLADE ASPECT RATIO-LAST STAGE	2.0000	3.0000
G	0.5000	COMPRESSOR EXIT MACH NUMBER	0.4500	0.5500
H	0.0000	MAX. COMP. INLET T- 0=CALC'D	0.0000	0.0000
I	0.0000	MAX. COMP. EXIT T- 0=CALC'D	0.0000	0.0000
J	1.0000	MAX. SPEED RATIO RPMMAX/RPMD	1.0000	1.0000
K	0.0000	BLADE MATERIAL DENSITY 0=T-SET	0.0000	0.0000
L	2.	1=CONST.HUB,2=MEAN,3=TIP DESH.	0.	0.
M	1.0000	RPM SCALER TO MATCH KNOWN RPM	1.0000	1.0000
N	0.0000	TEMP FOR MATERIAL CHANGE	0.0000	0.0000
O	0.0000	WEIGHT SCALER 0=AS CALCULATED	0.0000	0.0000
P	1.8000	STATOR BLADE TAPER RATIO	0.0000	0.0000
S	0.0550	BLADE VOLUME RATIO	0.0000	0.0000

ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER

Q QUIT PROCESSING ENTIRE ENGINE
R REVIEW UPDATED VALUES
T TERMINATE - GO ON TO NEXT COMPONENT

T
COMPONENT 3 IS A SPLITTER

WEIGHT AND LENGTH ARE IGNORED UNLESS FIRST COMPONENT IN THE ENGINE (AFTER INLET)
IF INNER STREAM IS NOT PRIMARY ENTER A "1" OR ELSE JUST HIT RETURN

—
COMPONENT 6 IS A COMPRESSOR, OPTIONS ARE (ENTER CORRECT LETTER)

A- TYPICAL FAN
B- OUTER PORTION OF NON-ROTATING SPLITTER FAN
C- INNER PORTION OF NON-ROTATING SPLITTER FAN
D- OUTER PORTION OF ROTATING SPLITTER FAN
E- INNER PORTION OF ROTATING SPLITTER FAN
F- LOW PRESSURE COMPRESSOR
G- HIGH PRESSURE COMPRESSOR
F

INDICATE WHETHER THERE ARE STATORS OR IF THIS IS A CENTRIFUGAL COMPRESSOR

A- CALCULATE STATOR WEIGHT
B- NO STATOR WEIGHT
C- CENTRIFUGAL COMPRESSOR
A

—
INDICATE WHETHER THERE IS A FRONT FRAME IN THE COMPRESSOR

A- NO FRAME
B- SNGL BEARING FRAME FOR TFS AND TJS WITHOUT POWER TAKEOFF (PTO)
C- SINGLE BEARING FRAME WITH PTO
D- TWO BEARING FRAME WHICH EXTENDS OUTWARD TO THE FAN OUTER CASE AND HOLDS TWO BEARINGS WITH PTO

^A INDICATE WHETHER THERE IS A REAR FRAME IN THE COMPRESSOR
 A- NO FRAME
 B- SINGL BEARING FRAME FOR TFS AND TJS WITHOUT POWER TAKEOFF (PTO)
 C- SINGLE BEARING FRAME WITH PTO
 D- TWO BEARING FRAME WHICH EXTENDS OUTWARD TO THE FAN OUTER CASE AND HOLDS TWO BEARINGS WITH PTO
^A
 GEAR BOX INDICATOR, 0= NO GEAR BOX, N=SHAFT NUMBER FOR GEAR BOX

⁰
 YOU MAY SPECIFY THE NUMBER OF STAGES-OVERRIDES MAX PR/STG., OTHERWISE ENTER 0

⁰
 FAN/COMPRESSOR
 CODE VALUE DESCRIPTION LOW VALUE-HIGH VALUE
 A 0.5000 FACE INLET MACH NUMBER 0.4500 0.6000
 B 1.50000 MAX 1ST STAGE PRATIO 1.50000 1.80000
 C 0.400 COMPRESSOR FACE HUB TO TIP RATIO 0.400 0.500
 D 1.5000 BLADE SOLIDITY (CORD/SPACING) 1.0000 1.5000
 E 4.0000 BLADE ASPECT RATIO-1ST STAGE 3.0000 5.0000
 F 3.0000 BLADE ASPECT RATIO-LAST STAGE 2.0000 3.0000
 G 0.4500 COMPRESSOR EXIT MACH NUMBER 0.4500 0.5500
 H 0.0000 MAX. COMP. INLET T- 0=CALC'D 0.0000 0.0000
 I 0.0000 MAX. COMP. EXIT T- 0=CALC'D 0.0000 0.0000
 J 1.0000 MAX. SPEED RATIO RPMMAX/RPMID 1.0000 1.0000
 K 0.0000 BLADE MATERIAL DENSITY 0=T-SET 0.0000 0.0000
 L 2. 1=CONST.HUB,2=MEAN,3=Tip DESH. 0. 0.
 M 1.0000 RPM SCALER TO MATCH KNOWN RPM 1.0000 1.0000
 N 0.0000 TEMP FOR MATERIAL CHANGE 0.0000 0.0000
 O 0.0000 WEIGHT SCALER 0=AS CALCULATED 0.0000 0.0000
 P 1.2000 STATOR BLADE TAPER RATIO 0.0000 0.0000
 S 0.1200 BLADE VOLUME RATIO 0.0000 0.0000

ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER

Q QUIT PROCESSING ENTIRE ENGINE
 R REVIEW UPDATED VALUES
 T TERMINATE - GO ON TO NEXT COMPONENT

C .85

E 2.

F 2.

L 3.

T COMPONENT 7 IS A DUCT, OPTIONS ARE (ENTER CORRECT LETTER)

A- PRIMARY BURNER
 B- DUCT BURNER
 C- AUGMENTOR
 D- DUCT
 ENTER LETTER

^{-D}

TYPE OF DUCT
 A- DUMMY - NO WT OR LENGTH
 B- INPUT LENGTH (WILL SPECIFY L/D OF DUCT)
 C- LENGTH FROM CONNECTION BETWEEN SPLITTER AND MIXER
 D- CROSSOVER DUCT FOR CENTRIFUGAL COMPRESSORS
 E- DIFFUSER FOR CENTRIFUGAL COMPRESSORS

^{-B}

DUCT
 CODE VALUE DESCRIPTION LOW VALUE-HIGH VALUE
 A 0.40 DUCT MACH HUIRER 0.40 0.50

B 1.00000 LEN/HT IF MODE B 0.00000 0.00000
 C 0.000 DUCT MEAN DIAM. IF =0 CALC. PER NODE BELOW 0. 0.
 D -1. IF NODE,0=MEAN D SPECIFIED,-1=CONN. COMP. 0. 0.
 ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER
 Q QUIT PROCESSING ENTIRE ENGINE
 R REVIEW UPDATED VALUES
 T TERMINATE - GO ON TO NEXT COMPONENT

A .45
 B 5.5
 T

COMPONENT 8 IS A COMPRESSOR. OPTIONS ARE (ENTER CORRECT LETTER)

A- TYPICAL FAN
 B- OUTER PORTION OF NON-ROTATING SPLITTER FAN
 C- INNER PORTION OF NON-ROTATING SPLITTER FAN
 D- OUTER PORTION OF ROTATING SPLITTER FAN
 E- INNER PORTION OF ROTATING SPLITTER FAN
 F- LOW PRESSURE COMPRESSOR
 G- HIGH PRESSURE COMPRESSOR
 G

INDICATE WHETHER THERE ARE STATORS OR IF THIS IS A CENTRIFUGAL COMPRESSOR

A- CALCULATE STATOR WEIGHT
 B- NO STATOR WEIGHT
 C- CENTRIFUGAL COMPRESSOR
 A

INDICATE WHETHER THERE IS A FRONT FRAME IN THE COMPRESSOR

A- NO FRAME
 B- SHGL BEARING FRAME FOR TFS AND TJS WITHOUT POWER TAKEOFF (PTO)
 C- SINGLE BEARING FRAME WITH PTO
 D- TWO BEARING FRAME WHICH EXTENDS OUTWARD TO THE FAN OUTER CASE AND HOLDS TWO BEARINGS WITH PTO
 A

INDICATE WHETHER THERE IS A REAR FRAME IN THE COMPRESSOR

A- NO FRAME
 B- SHGL BEARING FRAME FOR TFS AND TJS WITHOUT POWER TAKEOFF (PTO)
 C- SINGLE BEARING FRAME WITH PTO
 D- TWO BEARING FRAME WHICH EXTENDS OUTWARD TO THE FAN OUTER CASE AND HOLDS TWO BEARINGS WITH PTO
 A

GEAR BOX INDICATOR,0= NO GEAR BOX, N=SHAFT NUMBER FOR GEAR BOX

0
 YOU MAY SPECIFY THE NUMBER OF STAGES-OVERRIDES MAX PR/STG., OTHERWISE ENTER 0

0
 FAN/COMPRESSOR
 CODE VALUE DESCRIPTION LOW VALUE-HIGH VALUE
 A 0.4000 FACE INLET MACH NUMBER 0.4000 0.5000
 B 1.40000 MAX 1ST STAGE PRATIO 1.40000 1.70000
 C 0.700 COMPRESSOR FACE HUB TO TIP RATIO 0.600 0.800
 D 1.5000 BLADE SOLIDITY (CORD/SPACING) 1.0000 1.5000
 E 3.0000 BLADE ASPECT RATIO-1ST STAGE 2.0000 5.0000
 F 1.5000 BLADE ASPECT RATIO-LAST STAGE 1.0000 2.0000
 G 0.3000 COMPRESSOR EXIT MACH NUMBER 0.2000 0.3000
 H 0.0000 MAX. COMP. INLET T- 0=CALC'D 0.0000 0.0000
 I 0.0000 MAX. COIP. EXIT T- 0=CALC'D 0.0000 0.0000
 J 1.0000 MAX. SPEED RATIO RPM/IMAX/RPMID 1.0000 1.0000
 K 0.0000 BLADE MATERIAL DENSITY 0=T-SET 0.0000 0.0000
 L 2. 1=CONST.HUB,2=MEAN,3=TIP DESH. 0. 0.
 -M 1.0000 RPM SCALER TO MATCH KNOWN RPM 1.0000 1.0000
 N 0.0000 TEMP FOR MATERIAL CHANGE 0.0000 0.0000
 O 0.0000 WEIGHT SCALER 0=AS CALCULATED 0.0000 0.0000
 P 1.2000 STATOR BLADE TAPER RATIO 0.0000 0.0000

S 0.1200 BLADE VOLUME RATIO 0.0000 0.0000
ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER

Q QUIT PROCESSING ENTIRE ENGINE

R REVIEW UPDATED VALUES

T TERMINATE - GO ON TO NEXT COMPONENT

A .45

B 1.31

C .47

E 4.5

L 3.

T

COMPONENT 17 IS A DUCT, OPTIONS ARE (ENTER CORRECT LETTER)

A- PRIMARY BURNER

B- DUCT BURNER

C- AUGMENTOR

D- DUCT

ENTER LETTER

D

TYPE OF DUCT

A- DUMMY - NO WT OR LENGTH

B- INPUT LENGTH (WILL SPECIFY L/D OF DUCT)

C- LENGTH FROM CONNECTION BETWEEN SPLITTER AND MIXER

D- CROSSOVER DUCT FOR CENTRIFUGAL COMPRESSORS

E- DIFFUSER FOR CENTRIFUGAL COMPRESSORS

B DUCT

CODE	VALUE	DESCRIPTION	LOW	VALUE	HIGH	VALUE
------	-------	-------------	-----	-------	------	-------

A 0.40 DUCT MACH NUMBER 0.40 0.50

B 1.00000 LEN/HT IF NODE B 0.00000 0.00000

C 0.000 DUCT MEAN DIAM. IF =0 CALC. PER NODE BELOW 0. 0.

D -1. IF NODE,0=MEAN D SPECIFIED,-1=CONN. COMP. 0. 0.

ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER

Q QUIT PROCESSING ENTIRE ENGINE

R REVIEW UPDATED VALUES

T TERMINATE - GO ON TO NEXT COMPONENT

A .3

B 8.9

T

COMPONENT 9 IS A DUCT, OPTIONS ARE (ENTER CORRECT LETTER)

A- PRIMARY BURNER

B- DUCT BURNER

C- AUGMENTOR

D- DUCT

ENTER LETTER

A

A- BURNER HAS A FRAME

B- NO FRAME

A PRIMARY BURNER

CODE	VALUE	DESCRIPTION	LOW	VALUE	HIGH	VALUE
------	-------	-------------	-----	-------	------	-------

A 100.00 BURNER THRU-FLOW VELOCITY 100.00 150.00

B 0.01500 AIRFLOW RESIDENCY TIME 0.01000 0.02000

C 0.000 MEAN DIAMETER - IF 0 MATCH' UPSTREAM CMPT. 0. 0.

D 0. CIPT NUMBER FOR MATCHING DIAMETER 0. 0.

E 0. NUMBER OF CANS FOR CAN BURNER 0. 0.

ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER

Q QUIT PROCESSING ENTIRE ENGINE
R REVIEW UPDATED VALUES
T TERMINATE - GO ON TO NEXT COMPONENT
B .818
T
COMPONENT 10 IS A TURBINE, OPTIONS ARE (ENTER CORRECT LETTER)
H- HIGH PRESSURE TURBINE
L- LOW PRESSURE TURBINE
H
A- NO FRAME
B- TURBINE HAS EXIT FRAME
A
ENTER COMPONENT NUMBER OF COMPRESSOR THAT SETS TURBINE RPM
08
ENTER COMPONENT NUMBER THAT LIMITS TURBINE OUTER RADIUS
(+ = OUTLET, - = INLET, 0 = FEEDING COMPONENT)
-08
YOU MAY ENTER THE NUMBER OF STAGES IF DESIRED, OTHERWISE ENTER 0

11
R- RADIAL FLOW TURBINE
A- AXIAL FLOW TURBINE
A
TURBINE
CODE VALUE DESCRIPTION LOW VALUE-HIGH VALUE
A 0.3000 FACE INLET MACH NUMBER 0.3000 0.4000
B 0.28000 LOADING PARAMETER 0.20000 0.30000
C 1.500 BLADE SOLIDITY (CORD/SPACING) 1.000 1.500
D 1.5000 BLADE ASPECT RATIO- 1ST STAGE 1.0000 2.0000
E 1.5000 BLADE ASPECT RATIO-LAST STAGE 1.0000 2.0000
F 0.4500 TURBINE EXIT MACH NUMBER 0.4500 0.5000
G 125000. DISC REFERENCE STRESS 100000. 150000.
H 2. 1=CONST TIP DESH,2=MEAN,3=HUB 0. 0.
I 1.0000 MAX. SPEED RATIO RPMMAX/RPM0 1.0000 1.0000
J 0.0000 CONTROL RADIUS- 0 IF TRANSFRD 0.0000 0.0000
K 0.0000 BLADE MATERIAL DENSITY 0=T-SET 0.0000 0.0000
L 0.1550 BLADE VOLUME FACTOR 0.0000 0.0000
P 1.0000 BLADE TAPER RATIO 0.0000 0.0000
S 0.1550 STATOR BLADE VOLUME FACTOR 0.0000 0.0000
ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER

Q QUIT PROCESSING ENTIRE ENGINE
R REVIEW UPDATED VALUES
T TERMINATE - GO ON TO NEXT COMPONENT
A .4
B .2
F .5
G 150000.
H 3.
T
COMPONENT 11 IS A DUCT, OPTIONS ARE (ENTER CORRECT LETTER)

A- PRIMARY BURNER
B- DUCT BURNER
C- AUGMENTOR
D- DUCT
ENTER LETTER
D
TYPE OF DUCT
A- DUMMY - NO WT OR LENGTH

B- INPUT LENGTH (WILL SPECIFY L/D OF DUCT)
 C- LENGTH FROM CONNECTION BETWEEN SPLITTER AND MIXER
 D- CROSSOVER DUCT FOR CENTRIFUGAL COMPRESSORS
 E- DIFFUSER FOR CENTRIFUGAL COMPRESSORS

B
 DUCT
 CODE VALUE DESCRIPTION LOW VALUE-HIGH VALUE
 A 0.40 DUCT MACH NUMBER 0.40 0.50
 B 1.00000 LEN/HT IF NODE B 0.00000 0.00000
 C 0.000 DUCT MEAN DIAM. IF =0 CALC. PER NODE BELOW 0. 0.
 D -1. IF NODE,0=MEAN D SPECIFIED,-1=CONN. COMP. 0. 0.
 ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER
 Q QUIT PROCESSING ENTIRE ENGINE
 R REVIEW UPDATED VALUES
 T TERMINATE - GO ON TO NEXT COMPONENT

A .5
 B 3.5

T
 COMPONENT 12 IS A TURBINE, OPTIONS ARE (ENTER CORRECT LETTER)
 H- HIGH PRESSURE TURBINE

L- LOW PRESSURE TURBINE

L
 A- NO FRAME

B- TURBINE HAS EXIT FRAME

B
 ENTER COMPONENT NUMBER OF COMPRESSOR THAT SETS TURBINE RPM

02
 ENTER COMPONENT NUMBER THAT LIMITS TURBINE OUTER RADIUS
 (+ = OUTLET, - = INLET,0 = FEEDING COMPONENT)

+36
 YOU MAY ENTER THE NUMBER OF STAGES IF DESIRED, OTHERWISE ENTER 0

0
 R- RADIAL FLOW TURBINE
 A- AXIAL FLOW TURBINE
 A
 TURBINE
 CODE VALUE DESCRIPTION LOW VALUE-HIGH VALUE
 A 0.4500 FACE INLET MACH NUMBER 0.4000 0.5000
 B 0.28000 LOADING PARAMETER 0.20000 0.30000
 C 1.500 BLADE SOLIDITY (CORD/SPACING) 1.000 1.500
 D 2.0000 BLADE ASPECT RATIO- 1ST STAGE 2.0000 3.0000
 E 4.0000 BLADE ASPECT RATIO-LAST STAGE 4.0000 6.0000
 F 0.5500 TURBINE EXIT MACH NUMBER 0.5500 0.6000
 G 125000. DISC REFERENCE STRESS 100000. 150000.
 H 2. 1=CONST TIP DESN,2=MEAN,3=HUB 0. 0.
 I 1.0000 MAX. SPEED RATIO RPNMAX/RPND 1.0000 1.0000
 J 0.0000 CONTROL RADIUS- 0 IF TRANSFRD 0.0000 0.0000
 K 0.0000 BLADE MATERIAL DENSITY 0=T-SET 0.0000 0.0000
 L 0.1950 BLADE VOLUME FACTOR 0.0000 0.0000
 P 1.0000 BLADE TAPER RATIO 0.0000 0.0000
 S 0.1950 STATOR BLADE VOLUME FACTOR 0.0000 0.0000
 ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER

Q QUIT PROCESSING ENTIRE ENGINE
 R REVIEW UPDATED VALUES
 T TERMINATE - GO ON TO NEXT COMPONENT

A .5
 B .3

F .6
G 150000.
H 1.

T

COMPONENT 13 IS A DUCT, OPTIONS ARE (ENTER CORRECT LETTER)

- A- PRIMARY BURNER
- B- DUCT BURNER
- C- AUGMENTOR
- D- DUCT

ENTER LETTER

D

TYPE OF DUCT

- A- DUMMY - NO WT OR LENGTH
- B- INPUT LENGTH (WILL SPECIFY L/D OF DUCT)
- C- LENGTH FROM CONNECTION BETWEEN SPLITTER AND MIXER
- D- CROSSOVER DUCT FOR CENTRIFUGAL COMPRESSORS
- E- DIFFUSER FOR CENTRIFUGAL COMPRESSORS

A

DUCT

CODE	VALUE	DESCRIPTION	LOW VALUE	HIGH VALUE
A	0.40	DUCT MACH NUMBER	0.40	0.50
B	1.00000	LEN/HT IF MODE B	0.00000	0.00000
C	0.000	DUCT MEAN DIAM. IF =0 CALC. PER NODE BELOW	0.	0.
D	-1.	IF NODE, 0=MEAN D SPECIFIED, -1=CONN. COMP.	0.	0.

ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER
Q QUIT PROCESSING ENTIRE ENGINE
R REVIEW UPDATED VALUES
T TERMINATE - GO ON TO NEXT COMPONENT

A .6

B .5

T

COMPONENT 14 IS A NOZZLE, ENTER NOZZLE TYPE

- 1= CONVERGENT
- 2= C-D VARIABLE AREA

I

ENTER COMPONENT NUMBER THAT LIMITS NOZZLE OUTER RADIUS
(+ = OUTLET, - = INLET, 0 = FEEDING COMPONENT)

0

ENTER THRUST REVERSER TYPE, 0=NONE, 1=FAN, 2=PRIMARY

O

NOZZLE

CODE	VALUE	DESCRIPTION	LOW VALUE	HIGH VALUE
A	1.0000	LENGTH TO DIAMETER RATIO	1.0000	2.0000
B	0.0000	BYPASS RATIO FOR MIXED FLOW	0.0000	0.0000

ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER
Q QUIT PROCESSING ENTIRE ENGINE
R REVIEW UPDATED VALUES
T TERMINATE - GO ON TO NEXT COMPONENT

T

COMPONENT 4 IS A DUCT, OPTIONS ARE (ENTER CORRECT LETTER)

- A- PRIMARY BURNER
- B- DUCT BURNER
- C- AUGMENTOR
- D- DUCT

ENTER LETTER

D

TYPE OF DUCT
 A- DUMMY - NO WT OR LENGTH
 B- INPUT LENGTH (WILL SPECIFY L/D OF DUCT)
 C- LENGTH FROM CONNECTION BETWEEN SPLITTER AND MIXER
 D- CROSSOVER DUCT FOR CENTRIFUGAL COMPRESSORS
 E- DIFFUSER FOR CENTRIFUGAL COMPRESSORS

B
 DUCT
 CODE VALUE DESCRIPTION LOW VALUE-HIGH VALUE
 A 0.40 DUCT MACH NUMBER 0.40 0.50
 B 1.00000 LEN/HT IF MODE B 0.00000 0.00000
 C 0.000 DUCT MEAN DIAM. IF =0 CALC. PER NODE BELOW 0. 0.
 D -1. IF NODE,0=MEAN D SPECIFIED,-1=CONN. COMP. 0. 0.
 ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER
 Q QUIT PROCESSING ENTIRE ENGINE
 R REVIEW UPDATED VALUES
 T TERMINATE - GO ON TO NEXT COMPONENT
 A .5
 T

COMPONENT 5 IS A NOZZLE, ENTER NOZZLE TYPE

1= CONVERGENT
 2= C-D VARIABLE AREA

1
 ENTER COMPONENT NUMBER THAT LIMITS NOZZLE OUTER RADIUS
 (+ = OUTLET, - = INLET, 0 = FEEDING COMPONENT)

0
 ENTER THRUST REVERSER TYPE, 0=NONE, 1=FAN, 2=PRIMARY

0
 NOZZLE
 CODE VALUE DESCRIPTION LOW VALUE-HIGH VALUE
 A 1.0000 LENGTH TO DIAMETER RATIO 1.0000 2.0000
 B 0.0000 BYPASS RATIO FOR MIXED FLOW 0.0000 0.0000
 ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER
 Q QUIT PROCESSING ENTIRE ENGINE
 R REVIEW UPDATED VALUES
 T TERMINATE - GO ON TO NEXT COMPONENT
 T

COMPONENT 15 IS A SHAFT, ENTER SHAFT NUMBER FROM INNER TO OUTER, I.E. 1 2 3 ETC.

T
 ENTER TURBINES CONNECTED TO THIS SHAFT

12
 ENTER COMPONENT NUMBER OF FIRST COMPRESSOR ON THIS SHAFT

02
 SHAFT
 CODE VALUE DESCRIPTION LOW VALUE-HIGH VALUE
 A 50000.00 SHAFT ALLOWABLE STRESS 40000.00 50000.00
 B 0.29 SHAFT MATERIAL DENSITY 0.28 0.31
 C 0.0000 DIAMETER RATIO (INNER/OUTER) 0.0000 0.8500
 D 0. COMPONENT NUMBER ON SHAFT FOR INERTIA 0. 0.
 E 0. COMPONENT NUMBER ON SHAFT FOR INERTIA 0. 0.
 F 0. COMPONENT NUMBER ON SHAFT FOR INERTIA 0. 0.
 G 0. COMPONENT NUMBER ON SHAFT FOR INERTIA 0. 0.

ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POINT) OR ENTER
 Q QUIT PROCESSING ENTIRE ENGINE

APPENDIX B

Card Images from PREPWRITE AND NNEP/WATE Input Dataset

```
!W IPLT= T,PLOT= T,ISII= F,ISIO= F,
INT=4,IOUTCD=2,
DISKWI=0.,ACCS=0.1,ISCALE(1)=2,3,SCALE(1)=1.,.8,1.2,ACCARM=0,
ILENG(1)= 2, 3, 6, 7, 8,17, 9,10,11,12,13,14,
IWMEC(1, 2)=4HFAN , 0, 0, 4, 0, 0, 0,
IWMEC(1, 3)=4HSPLT, 0, 0, 0, 0, 0, 0,
IWMEC(1, 4)=4HDUCT, 2, 0, 0, 0, 0, 0,
IWMEC(1, 5)=4HNOZ , 1, 0, 0, 0, 0, 0,
IWMEC(1, 6)=4HLPC , 1, 0, 0, 0, 0, 0,
IWMEC(1, 7)=4HDUCT, 2, 0, 0, 0, 0, 0,
IWMEC(1, 8)=4HHPCE, 1, 0, 0, 0, 0, 0,
IWMEC(1, 9)=4HPBUR, 1, 0, 0, 0, 0, 0,
INMEC(1,10)=4HHPT , 0, 5, -8, 0, 0, 0,
IWMEC(1,11)=4HDUCT, 2, 0, 0, 0, 0, 0,
IWMEC(1,12)=4HLPT , 1, 2, 6, 0, 0, 0,
IWMEC(1,13)=4HDUCT, 1, 0, 0, 0, 0, 0,
IWMEC(1,14)=4HNOZ , 1, 0, 0, 0, 0, 0,
IWMEC(1,15)=4HSHAF, 1, 12, 0, 0, 0, 0,
IWMEC(1,16)=4HSHAF, 2, 10, 0, 0, 0, 0,
IWMEC(1,17)=4HDUCT, 2, 0, 0, 0, 0, 0,
DESV(1,2)=0.5500,1.800,0.3700,1.500,4.500,4.500,0.5000,
0.0000,0.0000,1.000,0.0000,2.000,1.000,0.0000,0.0000,1.800,0.5500D-01,
DESV(1,3)=0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESV(1,4)=0.5000,1.000,0.0000,-1.000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESV(1,5)=1.000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESV(1,6)=0.5000,1.500,0.8500,1.500,2.000,2.000,0.4500,
0.0000,0.0000,1.000,0.0000,3.000,1.000,0.0000,0.0000,1.200,0.1200,
DESV(1,7)=0.4500,5.500,0.0000,-1.000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESV(1,8)=0.4500,1.310,0.4700,1.500,4.500,1.500,0.3000,
0.0000,0.0000,1.000,0.0000,3.000,1.000,0.0000,0.0000,1.200,0.1200,
DESV(1,9)=100.0,0.1800D-01,0.0000,0.0000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESV(1,10)=0.4000,0.2000,1.500,1.500,0.5000,0.1500D06,
3.000,1.000,0.0000,0.0000,0.1500,0.0000,0.0000,0.0000,1.000,0.1550,
DESV(1,11)=0.5000,3.500,0.0000,-1.000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESV(1,12)=0.5000,0.3000,1.500,2.000,4.000,0.6000,0.1500D06,
1.000,1.000,0.0000,0.0000,0.1950,0.0000,0.0000,0.0000,1.000,0.1950,
DESV(1,13)=0.6000,0.5000,0.0000,-1.000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESV(1,14)=1.000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESV(1,15)=0.5000D05,0.3000,0.8500,2.000,6.000,12.00,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESV(1,16)=0.5000D05,0.3000,0.0000,8.000,10.00,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESV(1,17)=0.3000,8.900,0.0000,-1.000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
&END
```

SAMPLE CASE FOR PREPWATE
 &D
 &END
 &D MODE=1,
 IWT=4,
 KOHFIG(1,1)='INLT',1,0,2,0,SPEC(1,1)=1000.,4*0,1.,2*0,.80001,2*0,20.,
 KOHFIG(1,2)='COMP',2,0,3,0,SPEC(1,2)=1.8,0,1,3761,1,3762,1,3763,1,0,0,.824,
 1.734,1
 KOHFIG(1,3)='SPLT',3,0,7,4,SPEC(1,3)=4,323,
 KOHFIG(1,4)='DUCT',4,0,5,0,SPEC(1,4)=6*0,
 KOHFIG(1,5)='HOZZ',5,0,6,0,SPEC(1,5)=0,1,0,0,.99,0,0,0,1,
 KOHFIG(1,6)='COMP',7,0,8,0,SPEC(1,6)=1.3,0,1,3761,1,3762,1,3763,1,0,0,.867,
 1.373,1
 KOHFIG(1,7)='DUCT',8,0,9,0,SPEC(1,7)=6*0,
 KOHFIG(1,8)='COMP',9,0,10,17,SPEC(1,8)=1.3,.162,1,3707,1,3708,1,3709,1,0,0,
 .867,12,75,.985,
 KOHFIG(1,9)='DUCT',19,0,11,0,SPEC(1,9)=.048,0,0,2990.,1.,18400.,
 KOHFIG(1,10)='TURB',11,17,12,0,SPEC(1,10)=4,.79,1,3801,1,3802,1,1,.625,
 1,.919,5*0,1,
 KOHFIG(1,11)='DUCT',12,0,13,0,SPEC(1,11)=6*0,
 KOHFIG(1,12)='TURB',13,17,14,0,SPEC(1,12)=2.5,.21,1,3803,1,3804,1,1,0,1,.912,
 5244,1
 KOHFIG(1,13)='DUCT',14,0,15,0,SPEC(1,13)=6*0,
 KOHFIG(1,14)='HOZZ',15,0,16,0,SPEC(1,14)=0,1,0,0,.99,0,0,0,1,
 KOHFIG(1,15)='SHFT',2,6,12,0,SPEC(1,15)=4000,8*1
 KOHFIG(1,16)='SHFT',8,10,0,0,SPEC(1,16)=6000,8*1,
 KOHFIG(1,17)='DUCT',10,0,19,0,SPEC(1,17)=6*0,
 &END
 &N IPLOT= T,PLT= T,ISII= F,ISIO= F,
 IWT=4,IOUTCD=2,
 DISKWI=0.,ACCS=0.1,ISCALE(1)=2,3,SCALE(1)=1.,.8,1.2,ACCARM=0,
 ILENG(1)= 2, 3, 6, 7, 8,17, 9,10,11,12,13,14,
 IUMEC(1, 2)=4HFAN, 0, 0, 4, 0, 0, 0, 0,
 IUMEC(1, 3)=4HSPLT, 0, 0, 0, 0, 0, 0, 0,
 IUMEC(1, 4)=4IDUCT, 2, 0, 0, 0, 0, 0, 0,
 IUMEC(1, 5)=4IHOZ, 1, 0, 0, 0, 0, 0, 0,
 IUMEC(1, 6)=4ILPBC, 1, 0, 0, 0, 0, 0, 0,
 IUMEC(1, 7)=4IDUCT, 2, 0, 0, 0, 0, 0, 0,
 IUMEC(1, 8)=4IHPBC, 1, 0, 0, 0, 0, 0, 0,
 IUMEC(1, 9)=4IPBUR, 1, 0, 0, 0, 0, 0, 0,
 IUMEC(1,10)=4IHPPT, 0, 8, -8, 0, 0, 0, 0,
 IUMEC(1,11)=4IDUCT, 2, 0, 0, 0, 0, 0, 0,
 IUMEC(1,12)=4ILPCT, 1, 2, 6, 0, 0, 0, 0,
 IUMEC(1,13)=4IDUCT, 1, 0, 0, 0, 0, 0, 0,
 IUMEC(1,14)=4IHOZ, 1, 0, 0, 0, 0, 0, 0,
 IUMEC(1,15)=4ISHAF, 1, 12, 0, 0, 0, 0, 2,
 IUMEC(1,16)=4ISHAF, 2, 10, 0, 0, 0, 0, 0,
 IUMEC(1,17)=4IDUCT, 2, 0, 0, 0, 0, 0, 0,
 - DESVAL(1,2)=0.5500,1.800,0.3700,1.500,4.500,0.5000,
 0.0000,0.0000,1.000,0.0000,2.000,1.000,0.0000,0.0000,1.800,0.5500D-01,
 DESVAL(1,3)=0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
 0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
 DESVAL(1,4)=0.5000,1.000,0.0000,-1.000,0.0000,0.0000,0.0000,
 0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
 DESVAL(1,5)=1.000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
 0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
 - DESVAL(1,6)=0.5000,1.500,0.8500,1.500,2.000,2.000,0.4500,
 0.0000,0.0000,1.000,0.0000,3.000,1.000,0.0000,0.0000,1.200,0.1200,
 DESVAL(1,7)=0.4500,5.500,0.0000,-1.000,0.0000,0.0000,0.0000,
 0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,

```
DESV(1,8)=0.4500,1.310,0.4700,1.500,4.500,1.500,0.3000,  
0.0000,0.0000,1.000,0.0000,3.000,1.000,0.0000,0.0000,1.200,0.1200,  
DESV(1,9)=100.0,0.18000D-01,0.0000,0.0000,0.0000,0.0000,0.0000,  
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,  
DESV(1,10)=0.4000,0.2000,1.500,1.500,1.500,0.5000,0.1500D06,  
3.000,1.000,0.0000,0.0000,0.1550,0.0000,0.0000,0.0000,1.000,0.1550,  
DESV(1,11)=0.5000,3.500,0.0000,-1.000,0.0000,0.0000,0.0000,  
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,  
DESV(1,12)=0.5000,0.3000,1.500,2.000,4.000,0.6000,0.1500D06,  
1.000,1.000,0.0000,0.0000,0.1950,0.0000,0.0000,0.0000,1.000,0.1950,  
DESV(1,13)=0.6000,0.5000,0.0000,-1.000,0.0000,0.0000,0.0000,  
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,  
DESV(1,14)=1.000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,  
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,  
DESV(1,15)=0.5000D05,0.3000,0.8500,2.000,6.000,12.00,0.0000,  
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,  
DESV(1,16)=0.5000D05,0.3000,0.0000,8.000,10.00,0.0000,0.0000,  
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,  
DESV(1,17)=0.3000,8.900,0.0000,-1.000,0.0000,0.0000,0.0000,  
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,  
&END
```

APPENDIX C

Output from Sample Case from NNEP/MATE Code

SAMPLE CASE FOR PREPHATE
&D
&END

TABLE DATA INPUT SUMMARY 16 TABLES

TABLE NUMBER	REFERENCE NUMBER	ARRAY LOCATION
1	3761	1
2	3762	1075
3	3763	2149
4	3764	3223
5	3765	4297
6	3766	5371
7	3767	6445
8	3768	7681
9	3769	8917
10	3801	10153
11	3802	10606
12	3803	11203
13	3804	11656
14	3901	12397
15	3902	12799
16	3903	13213

DATA STORAGE ALLOCATION 20000
DATA STORAGE NOT USED 6385

```

&D MODE=1,
IWT=4,
KONFIG(1,1)='INLT',1,0,2,0,SPEC(1,1)=1000.,4x0,1.,2x0,.00001,2x0,20.,
KONFIG(1,2)='CDMP',2,0,3,0,SPEC(1,2)=1.8,0,1,3761,1,3762,1,3763,1,0,0,.824,
1.734,1,
KONFIG(1,3)='SPLT',3,0,7,4,SPEC(1,3)=4.323,
KONFIG(1,4)='DUCT',4,0,5,0,SPEC(1,4)=6x0,
KONFIG(1,5)='HOZZ',5,0,6,0,SPEC(1,5)=0,1,0,0,.99,0,0,0,1,
KONFIG(1,6)='COMP',7,0,8,0,SPEC(1,6)=1.3,0,1,3761,1,3762,1,3763,1,0,0,.867,
1.373,1,
KONFIG(1,7)='DUCT',8,0,9,0,SPEC(1,7)=6x0,
KONFIG(1,8)='COMP',9,0,10,17,SPEC(1,8)=1.3,.162,1,3707,1,3708,1,3709,1,0,0,
.867,12.75,.985,
KONFIG(1,9)='DUCT',19,0,11,0,SPEC(1,9)=.048,0,0,2990.,1.,18400.,
KONFIG(1,10)='TURB',11,17,12,0,SPEC(1,10)=4,.79,1,3801,1,3802,1,1,.625,
1,.919,5680,1,
KONFIG(1,11)='DUCT',12,0,13,0,SPEC(1,11)=6x0,
KONFIG(1,12)='TURB',13,17,14,0,SPEC(1,12)=2.5,.21,1,3803,1,3804,1,1,0,1,.912,
5244,1,
KONFIG(1,13)='DUCT',14,0,15,0,SPEC(1,13)=6x0,
KONFIG(1,14)='HOZZ',15,0,16,0,SPEC(1,14)=0,1,0,0,.99,0,0,0,1,
KONFIG(1,15)='SHFT',2,6,12,0,SPEC(1,15)=6000,8x1
KONFIG(1,16)='SHFT',8,10,0,0,SPEC(1,16)=6000,8x1,
KONFIG(1,17)='DUCT',10,0,19,0,SPEC(1,17)=6x0,

```

THE FOLLOWING REPRESENTS THE CONFIGURATION FOR MODE= 1
SAMPLE CASE FOR PREPWRITE
CONFIGURATION DATA 19 STATIONS 17 COMPONENTS

COMPONENT NUMBER	NKIND	COMPONENT TYPE	UPSTREAM STATIONS	DOWNTREAM STATIONS
1	1	INLET	1 0	2 0
2	4	COMPRESR	2 0	3 0
3	7	SPLITTER	3 0	7 4

4	2	DUCT B	4	0	5
5	9	NOZZLE	5	0	6
6	4	COMPRESR	7	0	8
7	2	DUCT B	8	0	9
8	4	COMPRESR	9	0	10
9	2	DUCT B	10	0	11
10	5	TURBIENE	11	17	12
11	2	DUCT B	12	0	13
12	5	TURBIENE	13	17	14
13	2	DUCT B	14	0	15
14	9	NOZZLE	15	0	16
15	11	SHAFT	2	6	12
16	11	SHAFT	8	10	13
17	112	DUCT B	10	0	14

CASE IDENTIFICATION SAMPLE CASE FOR PREPWRITE

INPUT DATA

```

MODE      1 NOW BEING USED
SUM OF (ERRORS*2)=      0.00000
&W IPLT= T, PLOT= T, ISII= F, ISIO= F,
IWT=4, IOUTCD=2,
DISKWI=0., ACCS=0.1, ISCALE(1)=2,3, SCALE(1)=1..8,1.2, ACCARM=0,
ILENG(1)= 2, 3, 6, 7, 8,17, 9,10,11,12,13,14,
IINNEC(1, 2)=4HFAN, 0, 0, 4, 0, 0, 0,
IINNEC(1, 3)=4HSPLT, 0, 0, 0, 0, 0, 0,
IINNEC(1, 4)=4HDUCT, 2, 0, 0, 0, 0, 0,
IINNEC(1, 5)=4HNOZ, 1, 0, 0, 0, 0, 0,
IINNEC(1, 6)=4HILPC, 1, 0, 0, 0, 0, 0,
IINNEC(1, 7)=4HDUCT, 2, 0, 0, 0, 0, 0,
IINNEC(1, 8)=4HIPC, 1, 0, 0, 0, 0, 0,
IINNEC(1, 9)=4HPBUR, 1, 0, 0, 0, 0, 0,
IINNEC(1,10)=4HHPPT, 0, 8, -8, 0, 0, 0,
IINNEC(1,11)=4HDUCT, 2, 0, 0, 0, 0, 0,
IINNEC(1,12)=4HLPT, 1, 2, 6, 0, 0, 0,
IINNEC(1,13)=4HDUCT, 1, 0, 0, 0, 0, 0,
IINNEC(1,14)=4HNOZ, 1, 0, 0, 0, 0, 0,
IINNEC(1,15)=4HSHAF, 1, 12, 0, 0, 0, 2,
IINNEC(1,16)=4HSHAF, 2, 10, 0, 0, 0, 8,
IINNEC(1,17)=4HDUCT, 2, 0, 0, 0, 0, 0,
DESVAL(1,2)=0.5500,1.800,0.370,1.500,4.500,4.500,0.5000,
0.0000,0.0000,1.000,0.0000,2.000,1.000,0.0000,0.0000,1.800,0.5500D-01,
DESVAL(1,3)=0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,4)=0.5000,1.000,0.0000,-1.000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,5)=1.000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,6)=0.5000,1.500,0.8500,1.500,2.000,2.000,0.4500,
0.0000,0.0000,1.000,0.0000,3.000,1.000,0.0000,0.0000,1.200,0.1200,
DESVAL(1,7)=0.4500,5.500,0.0000,-1.000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,8)=0.4500,1.310,0.4700,1.500,4.500,1.500,0.3000,
0.0000,0.0000,1.000,0.0000,3.000,1.000,0.0000,0.0000,1.200,0.1200,
DESVAL(1,9)=100.0.0.1800D-01,0.0000,0.0000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,10)=0.4000,0.2000,1.500,1.500,1.500,0.5000,0.1500D06,
3.000,1.000,0.0000,0.0000,0.1550,0.0000,0.0000,0.0000,1.000,0.1550,
DESVAL(1,11)=0.5000,3.500,0.0000,-1.000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,12)=0.5000,0.3000,1.500,2.000,4.000,0.6000,0.1500D06,
1.000,1.000,0.0000,0.0000,0.1950,0.0000,0.0000,0.0000,1.000,0.1950,
DESVAL(1,13)=0.6000,0.5000,0.0000,-1.000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,14)=1.000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,15)=0.5000D05,0.3000,0.8500,2.000,6.000,12.00,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,16)=0.5000D05,0.3000,0.0000,8.000,10.00,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
DESVAL(1,17)=0.3000,8.900,0.0000,-1.000,0.0000,0.0000,0.0000,
0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,0.0000,
-----END
XXXXXXXXXXXXXX
X           X

```

x FAN 2 x
x x
xxxxxxxxxxxxx2

MAX CONDITIONS OCCUR AT

xx
ALT MH VALUE
PTOT 0. 0.000 14.7 LB/SQIN
TTOT 0. 0.000 538.7 DEG R
CWTN 0. 0.000 1019.1 LB/SEC
xx

DUCT

M NO VEL T TOT P TOT P STAT AREA GAM
0.550 608. 539. 2116. 1723. 25.8790 1.4002

UTIPMAX STRESS DEN W/AREA TR H/T
1591.8 45894.5 0.168 4.970 1.800 0.370

COMPRESSOR 2 MECHANICAL DESIGN

LOADING N STG DIAM U TIP C RPM C RPM MAX RPM
0.927 1.00 74.14 1562.0 4920.4 4828.2 4920.4

STAGE 1

WD WB WS WH WC CL RHOB RHOD AR
225. 389. 0. 0. 35. 5.2 0.168 0.168 4.50

PR DEL H MACH AREA R HUB R TIP HB UTIPMAX STR WEIGHT TIN TMAX STAGE I
1.7337 26.7 0.550 25.879 13.72 37.07 67 1591.8 45894. 649. 539. 539. 165288.

FRAME WT = 806.03

N STG WEIGHT LENGTH CENGRA INERTIA
1 1455.30 10.38 8.1 165287.8

DUCT

M NO VEL T TOT P TOT P STAT AREA GAM
0.550 667. 658. 3670. 2989. 16.4039 1.3977

PR AD EF PD TO HP
1.7340 0.8240 3669.5 649.8 37781.
HI HO WI CWT
128.75 155.45 1000.00 1019.09

xxxxxxxxxxxxx TOTAL COMP WEIGHT IS 1455.303

xxxxxxxxxxxxx
x x
x LPC 6 x
x x
xxxxxxxxxxxxx2

MAX CONDITIONS OCCUR AT

xx
ALT MH VALUE
-PTOT 0. 0.000 25.5 LB/SQIN
TTOT 0. 0.000 649.8 DEG R
CWTN 0. 0.000 121.3 LB/SEC
xxxxxxxxxxxxxxxxxxxxxxxxxxxxxxxxxxxx

DUCT
M NO VEL T TOT P TOT P STAT AREA GAM
0.500 609. 650. 3670. 3094. 3.2903 1.3977

UTIPMAX STRESS DEN W/AREA TR H/T
1001.0 7725.8 0.168 0.627 1.200 0.850

COMPRESSOR 6 MECHANICAL DESIGN

LOADING N STG DIAM U TIP C RPM C RPM MAX RPM
0.866 1.00 46.63 894.3 4920.4 4396.1 4920.4

STAGE 1
WD WB WS WH WC CL RHOB RHOD AR
71. 31. 31. 162. 17. 4.1 0.168 0.168 2.00
PR DEL H MACH AREA R HUB R TIP NB UTIPMAX SIR WEIGHT TIN TMAX STAGE I
1.3741 17.1 0.500 3.290 19.82 23.31 125 1001.0 7726. 314. 650. 650. 26483.

N STG WEIGHT LENGTH CENGRA INERTIA
1 313.76 4.09 4.1 26482.8

DUCT
M NO VEL T TOT P TOT P STAT AREA GAM
0.500 641. 721. 5038. 4250. 2.5254 1.3953

PR AD EF PO TO HP
1.3730 0.8670 5038.3 720.6 4542.
HI HO WI CWI
155.45 172.54 187.86 121.26

XXXXXXXXXXXXXX TOTAL COMP WEIGHT IS 313.764

XXXXXXXXXXXXX
X X
X DUCT 7 X
X X
XXXXXXXXXXXXX2

MAX CONDITIONS OCCUR AT
XXXXXXXXXXXXXX

ALT MH
PTOT 0. 0.000
TTOT 0. 0.000
XXXXXXXXXXXXXX

DUCT 7
RH= 20.40 RT= 23.27 LENG= 15.76
AREA= 2.731 RH0= .168
CAS WT INC WT WTOT
6.3043 5.5279 11.8322

XXXXXXXXXXXXX
X X
X HPC 8 X
X X
XXXXXXXXXXXXX2

MAX CONDITIONS OCCUR AT
XXXXXXXXXXXXXX

ALT	MN	VALUE								
PTOT	0.	0.000 35.0 LB/SQIN								
TTOT	0.	0.000 720.6 DEG R								
CWIN	0.	0.000 93.0 LB/SEC								
XXXXXXXXXXXXXXXXXXXXXXXXXXXXXX										
DUCT										
M	NO	VEL	T	TOT	P	TOT	P	STAT	AREA	GAM
0.450	580.	721.	5038.	4387.	2.7306	1.3953				
UTIPMAX	STRESS	DEN	W/AREA	TR	H/T					
1198.8	30837.8	0.168	1.204	1.200	0.470					

COMPRESSOR & MECHANICAL DESIGN

LOADING	H	STG	DIAM	U	TIP	C	RPM	C RPM	MAX RPM
0.842	14.00	25.35	1017.1	10838.5	9195.5	10838.5			

STAGE 1
 WD WB WS WH WC CL RHOB RHOD AR
 9. 24. 24. 13. 8. 3.5 0.168 0.168 4.50
 PR DEL H MACH AREA R HUB R TIP NB UTIPMAX STR WEIGHT TIN TMAX STAGE I
 1.3028 15.1 0.450 2.731 5.96 12.67 80 1198.8 30838. 77. 721. 721. 1372.

STAGE 2
 WD WB WS WH WC CL RHOB RHOD AR
 15. 14. 14. 10. 6. 2.8 0.168 0.168 4.27
 PR DEL H MACH AREA R HUB R TIP NB UTIPMAX STR WEIGHT TIN TMAX STAGE I
 1.2768 15.1 0.439 2.227 7.65 12.67 101 1198.8 25244. 60. 783. 783. 1397.

STAGE 3
 WD WB WS WH WC CL RHOB RHOD AR
 18. 9. 9. 8. 5. 2.3 0.168 0.168 4.04
 PR DEL H MACH AREA R HUB R TIP NB UTIPMAX STR WEIGHT TIN TMAX STAGE I
 1.2550 15.1 0.429 1.849 8.71 12.67 121 1198.8 20998. 50. 844. 844. 1448.

STAGE 4
 WD WB WS WH WC CL RHOB RHOD AR
 19. 7. 7. 7. 5. 2.0 0.168 0.168 3.81
 PR DEL H MACH AREA R HUB R TIP NB UTIPMAX STR WEIGHT TIN TMAX STAGE I
 1.2366 15.1 0.418 1.559 9.44 12.67 140 1198.8 17722. 43. 906. 906. 1439.

STAGE 5
 WD WB WS WH WC CL RHOB RHOD AR
 18. 5. 5. 6. 4. 1.8 0.168 0.168 3.58
 PR DEL H MACH AREA R HUB R TIP NB UTIPMAX STR WEIGHT TIN TMAX STAGE I
 1.2206 15.1 0.407 1.331 9.98 12.67 158 1198.8 15149. 38. 967. 967. 1382.

STAGE 6
 WD WB WS WH WC CL RHOB RHOD AR
 16. 4. 4. 6. 4. 1.6 0.168 0.168 3.35
 PR DEL H MACH AREA R HUB R TIP NB UTIPMAX STR WEIGHT TIN TMAX STAGE I
 1.2068 15.1 0.396 1.150 10.39 12.67 174 1198.8 13100. 34. 1027. 1027. 1324.

STAGE 7
 WD WB WS WH WC CL RHOB RHOD AR
 15. 3. 3. 5. 3. 1.5 0.168 0.168 3.12
 PR DEL H MACH AREA R HUB R TIP NB UTIPMAX STR WEIGHT TIN TMAX STAGE I
 1.1947 15.1 0.386 1.004 10.71 12.67 188 1198.8 11443. 31. 1088. 1088. 1260.

STAGE 8

WD WB WS WN WC CL RHOB RHOD AR
 14. 3. 3. 5. 3. 1.4 0.168 0.168 2.88
 PR DEL H MACH AREA R HUB R TIP HB UTIPMAX STR WEIGHT TIN TMAX STAGE I
 1.1840 15.1 0.375 0.885 10.96 12.67 200 1198.8 10088. 28. 1147. 1147. 1191.

STAGE 9
 WD WB WS WN WC CL RHOB RHOD AR
 25. 4. 4. 5. 3. 1.3 0.286 0.286 2.65
 PR DEL H MACH AREA R HUB R TIP HB UTIPMAX STR WEIGHT TIN TMAX STAGE I
 1.1744 15.1 0.364 0.787 11.16 12.67 209 1198.8 15265. 42. 1206. 1206. 2142.

STAGE 10
 WD WB WS WN WC CL RHOB RHOD AR
 23. 4. 4. 5. 3. 1.3 0.286 0.286 2.42
 PR DEL H MACH AREA R HUB R TIP HB UTIPMAX STR WEIGHT TIN TMAX STAGE I
 1.1659 15.1 0.354 0.704 11.33 12.67 215 1198.8 13673. 39. 1265. 1265. 2013.

STAGE 11
 WD WB WS WN WC CL RHOB RHOD AR
 22. 4. 4. 5. 3. 1.3 0.286 0.286 2.19
 PR DEL H MACH AREA R HUB R TIP HB UTIPMAX STR WEIGHT TIN TMAX STAGE I
 1.1582 15.1 0.343 0.635 11.47 12.67 217 1198.8 12333. 37. 1324. 1324. 1913.

STAGE 12
 WD WB WS WN WC CL RHOB RHOD AR
 21. 3. 3. 5. 3. 1.3 0.286 0.286 1.96
 PR DEL H MACH AREA R HUB R TIP HB UTIPMAX STR WEIGHT TIN TMAX STAGE I
 1.1513 15.1 0.332 0.577 11.59 12.67 215 1198.8 11198. 35. 1382. 1382. 1842.

STAGE 13
 WD WB WS WN WC CL RHOB RHOD AR
 20. 3. 3. 5. 3. 1.3 0.286 0.286 1.73
 PR DEL H MACH AREA R HUB R TIP HB UTIPMAX STR WEIGHT TIN TMAX STAGE I
 1.1450 15.1 0.321 0.527 11.68 12.67 208 1198.8 10230. 34. 1439. 1439. 1796.

STAGE 14
 WD WB WS WN WC CL RHOB RHOD AR
 19. 3. 3. 5. 3. 1.4 0.286 0.286 1.50
 PR DEL H MACH AREA R HUB R TIP HB UTIPMAX STR WEIGHT TIN TMAX STAGE I
 1.1392 15.1 0.311 0.484 11.77 12.67 197 1198.8 9399. 34. 1496. 1496. 1776.

H STG WEIGHT LENGTH CENGRA INERTIA
 14 581.92 24.74 13.9 22295.5

DUCT
 M NO VEL T TOT P TOT P STAT AREA GAM
 0.311 584. 1553. 64238. 60225. 0.4352 1.3475

— PR AD EF PO TO HP
 12.7500 0.8670 64237.9 1553.3 56017.
 HI HO WI CMI
 172.54 383.29 187.86 93.01

***** TOTAL COMP WEIGHT IS 581.918

 x
 x DUCT 17 x
 x x
 *****2

MAX CONDITIONS OCCUR AT
XXXXXXXXXXXXXXXXXXXXXX

ALT MN
PTOT 0. 0.000

TTOT 0. 0.000

XXXXXXXXXXXXXXXXXXXXXX

DUCT 17

RH= 11.90 RT= 12.60 LENGTH= 6.27

AREA= 0.376 RHO=.286

CAS WT	INC WT	WTOT
11.3975	10.7609	22.1584

XXXXXXXXXXXXXX

X X

X PBUR 9 X

X X

XXXXXXXXXXXXXX2

27

MAX CONDITIONS OCCUR AT
XXXXXXXXXXXXXXXXXXXXXX

ALT MN VALUE

PTOT 0. 0.000 446.1 LB/SQIN

TTOT 0. 0.000 1553.3 DEG R

CWTH 0. 0.000 9.0 LB/SEC

XXXXXXXXXXXXXXXXXXXXXX

BURNER NUMBER 9

RIN ROUT LENGTH MACH WSPEC

10.177 14.023 21.600 0.053 4.419

CAS WT	LIN WT	NOZ WT	INC WT	FRAME	WTOT
43.0	54.2	21.7	31.2	178.6	328.7

XXXXXXXXXXXXXX

X X

X HPT 10 X

X X

XXXXXXXXXXXXXX2

MAX CONDITIONS OCCUR AT
XXXXXXXXXXXXXXXXXXXXXX

ALT MN VALUE

PTOT 0. 0.000 424.7 LB/SQIN

TTOT 0. 0.000 2990.0 DEG R

CHOUT 0. 0.000 55.4 LB/SEC

XXXXXXXXXXXXXXXXXXXXXX

DUCT

M NO	VEL	T TOT	P TOT	P STAT	AREA	GAM
0.400	998.	2877.	61154.	55218.	0.4795	1.2912

UTIPMAX STRESS	DEN	W/AREA	TR	H/T
1060.5	10488.8	0.286	0.294	1.000

— TURBINE 10 MECHANICAL DESIGN

H/T H STG LOADING AREA

0.908 2.000 0.200 0.479

UT	RTIP	RHUB	DEL H	RPM	MAXRPM	TORQ
----	------	------	-------	-----	--------	------

1060.5 11.2 10.2 224.6 10838.5 10838.5 325775.

STAGE 1
 DISK BLADE VANE HWD CASE AR
 6.9 2.6 13.2 26.1 4.8 1.50
 PR DEL H MACH AREA R HUB R TIP NB MAXUTIP STR WEIGHT LENGTH STAGE I

XXXX WARNING FOLLOWING STAGE DESIGN LIMIT EXCEEDED XXXXX

STAGE LOADING IS 0.20 DES LIMIT ISO.28

XXSTAGE LOADING IS TOO HIGH INCREASE LOADING INPUT 1/TLP=LOADING XX
 1.9609 112.3 0.400 0.479 10.18 11.21 154 1060.5 10489. 53.65 2.40 634.

STAGE 2
 DISK BLADE VANE HWD CASE AR
 11.5 7.2 36.2 42.1 8.3 1.50
 PR DEL H MACH AREA R HUB R TIP NB MAXUTIP STR WEIGHT LENGTH STAGE I

XXXX WARNING FOLLOWING STAGE DESIGN LIMIT EXCEEDED XXXXX

STAGE LOADING IS 0.20 DES LIMIT ISO.28

XXSTAGE LOADING IS TOO HIGH INCREASE LOADING INPUT 1/TLP=LOADING XX
 2.1890 112.3 0.450 0.796 10.18 11.84 101 1120.1 17422. 105.35 3.88 1356.

H STG LENGTH WEIGHT CENGRA INERTIA
 2 6.28 159.00 5.0 1990.

DUCT

M NO VEL T TOT P TOT P STAT AREA GAM
 0.500 1065. 2096. 14225. 12113. 1.5459 1.3103

PR TR AD EF PO TO TO.1
 4.2991 1.3553 0.9190 14224.8 2122.4 2096.4
 H IN H OUT AREA FLOW HP
 777.55 552.92 5.23 176.28 56024.

XXXXXXXXXXXXXXXXXXXX TOTAL TURB WEIGHT IS 158.997

XXXXXXXXXXXXXX
 X X
 X DUCT 11 X
 X X
 XXXXXXXXXXXXXXX2

MAX CONDITIONS OCCUR AT
 XXXXXXXXXXXXXXXXXXXXXXX

ALT MH
 PTOT 0. 0.000
 TTOT 0. 0.000
 XXXXXXXXXXXXXXXXXXXXXXX

DUCT 11
 RH= 10.00 RT= 13.07 LEHG= 10.74
 AREA= 1.546 RHO=.286
 CAS WT INC WT WTOT
 4.6581 3.5643 8.2224

XXXXXXXXXXXXXX
 X X
 X LPT 12 X

XXXXXXXXXXXX2

MAX CONDITIONS OCCUR AT

XXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXX
 ALT MH VALUE
 PTOT 0. 0.000 98.8 LB/SQIN
 TTOT 0. 0.000 2096.5 DEG R
 CWOUT 0. 0.000 211.0 LB/SEC
 XXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXX

DUCT

M NO VEL T TOT P TOT P STAT AREA GAM
 0.500 1065. 2097. 14229. 12117. 1.5455 1.3103

UTIPMAX STRESS DEN W/AREA TR H/T
 899.1 6967.9 0.286 0.505 1.000 0.916

TURBINE 12 MECHANICAL DESIGN

H/T H STG LOADING AREA
 0.916 3.000 0.300 1.546
 UT RTIP RHUB DEL H RPM MAXRPM TORQ
 899.1 20.9 19.2 161.5 4920.4 4920.4 542172.

STAGE 1

DISK BLADE VANE HWD CASE AR
 30.8 13.7 68.5 119.2 11.7 2.00
 PR DEL H MACH AREA R HUB R TIP HB MAXUTIP STR WEIGHT LENGTH STAGE I
 1.5488 53.8 0.500 1.546 19.17 20.94 223 899.1 6968. 243.83 3.10 10705.

STAGE 2

DISK BLADE VANE HWD CASE AR
 38.3 18.9 94.4 105.2 11.2 3.00
 PR DEL H MACH AREA R HUB R TIP HB MAXUTIP STR WEIGHT LENGTH STAGE I
 1.6212 53.8 0.533 2.178 18.40 20.94 233 899.1 9819. 267.93 2.97 12920.

STAGE 3

DISK BLADE VANE HWD CASE AR
 45.1 32.8 164.1 103.5 12.8 4.00
 PR DEL H MACH AREA R HUB R TIP HB MAXUTIP STR WEIGHT LENGTH STAGE I
 1.7172 53.8 0.567 3.209 17.07 20.94 203 899.1 14469. 358.29 3.40 16289.

FRAME WT = 318.95

H STG LENGTH WEIGHT CENGRA INERTIA
 3 12.62 1189.01 8.2 39914.

DUCT

M NO VEL T TOT P TOT P STAT AREA GAM
 0.600 1089. 1525. 3295. 2609. 5.1741 1.3350

PR TR AD EF PO TO TO.1
 4.3186 1.3760 0.9120 3294.8 1523.7 1524.6
 H IN H OUT AREA FLON IP
 544.69 383.23 20.12 185.29 42328.

XXXXXXXXXXXX TOTAL TURB WEIGHT IS 1189.008

XXXXXXXXXXXXXX
 X X
 X DUCT 13 X
 X X
 XXXXXXXXXXXXXXX2

MAX CONDITIONS OCCUR AT
 XXXXXXXXXXXXXXXXXXXXXXX
 ALT MN
 PTOT 0. 0.000
 TTOT 0. 0.000
 XXXXXXXXXXXXXXXXXXXXXXX

XXXXXXXXXXXXXX
 X X
 X NOZ 14 X
 X X
 XXXXXXXXXXXXXXX2

MAX CONDITIONS OCCUR AT
 XXXXXXXXXXXXXXXXXXXXXXX
 ALT MN
 PTOT 0. 0.000
 TTOT 0. 0.000
 XXXXXXXXXXXXXXXXXXXXXXX

NOZZLE 14
 WEIGHT= 155.91 LENGTH= 41.876 TR WT= 0.00

XXXXXXXXXXXXXX
 X X
 X DUCT 4 X
 X X
 XXXXXXXXXXXXXXX2

MAX CONDITIONS OCCUR AT
 XXXXXXXXXXXXXXXXXXXXXXX
 ALT MN
 PTOT 0. 0.000
 TTOT 0. 0.000
 XXXXXXXXXXXXXXXXXXXXXXX
 DUCT 4
 RH= 22.33 RT= 33.92 LENGTH= 11.59
 AREA= 14.224 RHO=.168
 CAS WT INC WT WTOT
 7.1743 4.7223 11.8966

XXXXXXXXXXXXXX
 X X
 X NOZ 5 X
 X X
 XXXXXXXXXXXXXXX2

MAX CONDITIONS OCCUR AT
 XXXXXXXXXXXXXXXXXXXXXXX
 ALT MN
 PTOT 0. 0.000
 TTOT 0. 0.000

XXXXXXXXXXXXXXXXXXXX
NOZZLE 5
WEIGHT= 242.87 LENGTH= 67.836 TR WT= 0.00

XXXXXXXXXXXXXX
X X
X SHAF 15 X
X X
XXXXXXXXXXXXX2

MAX TORQUE CONDITION
XXXXXXXXXXXXXXXXXXXX

TORQUE

10.6

XXXXXXXXXXXXXXXXXXXX

SHAFT 15

DO	DI	LENG	DN	WT
6.89	5.86	89.48	0.86	277.62

TOTAL INERTIA OF THIS SPOOL IS 231684.

XXXXXXXXXXXXXX
X X
X SHAF 16 X
X X
XXXXXXXXXXXXX2

MAX TORQUE CONDITION
XXXXXXXXXXXXXXXXXXXX

TORQUE

9.3

XXXXXXXXXXXXXXXXXXXX

SHAFT 16

DO	DI	LENG	DN	WT
7.72	7.29	27.87	2.12	42.22

TOTAL INERTIA OF THIS SPOOL IS 24286.
THE DN VALUE OF 2.12 MILLION IS HIGH

XXXXXXXXXXXXXX
X X
X ACCS WT X
X X
XXXXXXXXXXXXX2

- ACCS WT= 440.164

WEIGHT INPUT DATA IN ENGL UNITS
WEIGHT OUTPUT DATA IN ENGL UNITS

ENGINE SCALING DATA

XXXXXXXXXXXXXXXXXXXX

SCALE FACTOR 1.00

XXXXXXXXXXXXXXXXXXXX

COMP NO	WT EST	COMP LEN	ACCU LEN	UPSTREAM RI	UPSTREAM RO	RADIUS RI	RADIUS RO	DOWNSTREAM RI	DOWNSTREAM RO	RADIUS RI	RADIUS RO	NSTAGE
1	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0
2	1455.	10.	10.	14.	37.	0.	0.	19.	34.	0.	0.	1
3	0.	0.	10.	19.	34.	0.	0.	19.	23.	23.	34.	0
4	12.	12.	22.	22.	34.	0.	0.	22.	34.	0.	0.	0
5	243.	68.	90.	0.	34.	0.	0.	0.	32.	0.	0.	0
6	314.	4.	14.	20.	23.	0.	0.	20.	23.	0.	0.	1
7	12.	16.	30.	20.	23.	0.	0.	20.	23.	0.	0.	0
8	582.	25.	55.	6.	13.	0.	0.	12.	13.	0.	0.	1
9	329.	22.	83.	10.	14.	0.	0.	10.	14.	0.	0.	0
10	159.	6.	89.	10.	11.	0.	0.	10.	13.	0.	0.	2
11	8.	11.	100.	10.	13.	0.	0.	10.	13.	0.	0.	0
12	1189.	13.	112.	19.	21.	0.	0.	14.	21.	0.	0.	3
13	0.	0.	112.	14.	21.	0.	0.	14.	21.	0.	0.	0
14	156.	42.	154.	0.	21.	0.	0.	0.	19.	0.	0.	0
15	278.	0.	89.	14.	37.	0.	0.	10.	13.	0.	0.	0
16	42.	0.	0.	20.	23.	12.	13.	0.	0.	0.	0.	0
17	22.	6.	61.	12.	13.	0.	0.	12.	13.	0.	0.	0

TOTAL BARE ENGINE WEIGHT= 4800. ACCESSORIES= 440.16 ESTIMATED TOTAL LENGTH= 154. ESTIMATED MAXIMUM RADIUS= 37.
ESTIMATED CENTER OF GRAVITY= .51.
ESTIMATED AIRFLOW SCALING EXPONENT IS 1.000

WEIGHT INPUT DATA IN ENGL UNITS
WEIGHT OUTPUT DATA IN ENGL UNITS

ENGINE SCALING DATA
XXXXXXXXXXXXXXXXXXXX

SCALE FACTOR 0.80
XXXXXXXXXXXXXXXXXXXX

COMP NO	WT EST	COMP NO	ACCU LEN	LEN	UPSTREAM RI	RI	UPSTREAM RO	RI	DOWNSTREAM RI	RI	DOWNSTREAM RO	RI	RO	NSTAGE
1	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0
2	1223.	9.	9.	12.	33.	0.	0.	0.	17.	31.	0.	0.	0.	1
3	0.	0.	9.	17.	31.	0.	0.	0.	17.	21.	21.	31.	0.	0
4	9.	10.	20.	20.	31.	0.	0.	0.	20.	31.	0.	0.	0.	0
5	197.	61.	81.	0.	31.	0.	0.	0.	0.	29.	0.	0.	0.	0
6	223.	4.	13.	18.	21.	0.	0.	0.	18.	21.	0.	0.	0.	1
7	8.	14.	27.	18.	21.	0.	0.	0.	18.	21.	0.	0.	0.	0
8	419.	22.	49.	5.	11.	0.	0.	0.	11.	11.	0.	0.	0.	1
9	268.	22.	76.	9.	13.	0.	0.	0.	9.	13.	0.	0.	0.	0
10	115.	6.	82.	9.	10.	0.	0.	0.	9.	12.	0.	0.	0.	2
11	6.	10.	92.	9.	12.	0.	0.	0.	9.	12.	0.	0.	0.	0
12	914.	11.	103.	17.	19.	0.	0.	0.	13.	19.	0.	0.	0.	3
13	0.	0.	103.	13.	19.	0.	0.	0.	13.	19.	0.	0.	0.	0
14	125.	37.	140.	0.	19.	0.	0.	0.	0.	17.	0.	0.	0.	0
15	237.	0.	82.	12.	33.	0.	0.	0.	9.	12.	0.	0.	0.	0
16	38.	0.	0.	18.	21.	11.	11.	0.	0.	0.	0.	0.	0.	0
17	16.	6.	55.	11.	11.	0.	0.	0.	11.	11.	0.	0.	0.	0

TOTAL BARE ENGINE WEIGHT= 3799. ACCESSORIES= 347.68 ESTIMATED TOTAL LENGTH= 140. ESTIMATED MAXIMUM RADIUS= 33.
ESTIMATED CENTER OF GRAVITY= 46.
ESTIMATED AIRFLOW SCALING EXPONENT IS 1.049

WEIGHT INPUT DATA IN ENGL UNITS
WEIGHT OUTPUT DATA IN ENGL UNITS

ENGINE SCALING DATA

XXXXXXXXXXXXXXXXXXXX
SCALE FACTOR 1.20
XXXXXXXXXXXXXXXXXXXX

COMP	WT	COMP	ACCU	UPSTREAM	RADIUS	DOWNSTREAM	RADIUS	NSTAGE	
NO	EST	LEN	LEN	RI	RO	RI	RO	RI	RO
1	0.	0.	0.	0.	0.	0.	0.	0.	0.
2	1707.	11.	11.	15.	61.	0.	0.	21.	38.
3	0.	0.	11.	21.	38.	0.	0.	21.	25.
4	16.	13.	24.	24.	37.	0.	0.	24.	37.
5	288.	74.	98.	0.	37.	0.	0.	0.	35.
6	415.	4.	16.	22.	26.	0.	0.	22.	26.
7	16.	17.	33.	22.	25.	0.	0.	22.	25.
8	766.	27.	60.	7.	14.	0.	0.	13.	14.
9	389.	22.	89.	11.	15.	0.	0.	11.	15.
10	208.	7.	96.	11.	12.	0.	0.	11.	14.
11	11.	12.	107.	11.	14.	0.	0.	11.	14.
12	1489.	14.	121.	21.	23.	0.	0.	16.	23.
13	0.	0.	121.	16.	23.	0.	0.	16.	23.
14	187.	46.	167.	0.	23.	0.	0.	0.	23.
15	316.	0.	96.	15.	41.	0.	0.	11.	14.
16	46.	0.	0.	22.	25.	13.	14.	0.	0.
17	29.	7.	67.	13.	14.	0.	0.	13.	14.

TOTAL BARE ENGINE WEIGHT= 5882. ACCESSORIES= 540.64 ESTIMATED TOTAL LENGTH= 167. ESTIMATED MAXIMUM RADIUS= 41.
ESTIMATED CENTER OF GRAVITY= 56.
ESTIMATED AIRFLOW SCALING EXPONENT IS 1.114

UPDATED INPUT DATA TO REFLECT CALCULATED INPUT

COMPONENT	NO.	TYPE	DATINP1	DATINP2	DATINP3	DATINP4	DATINP5	DATINP6	DATINP7	DATINP8	DATINP9
1	1	INLET	0.10000D 04	0.00000	0.14696D 02	0.00000	0.00000	0.10000D 01	0.00000	0.00000	0.10000D 04
2	2	COMPRESR	0.18000D 01	0.00000	0.39250D 04	0.37610D 04	0.10227D 04	0.37620D 04	0.95772D 00	0.37630D 04	0.39127D 00
3	3	SPLITTER	0.43230D 01	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
4	4	DUCT B	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
5	5	NOZZLE	0.15353D 04	0.10000D 01	0.00000	0.00000	0.99000D 00	0.99000D 00	0.00000	0.00000	0.10000D 01
6	6	COMPRESR	0.13000D 01	0.00000	0.35737D 04	0.37610D 04	0.12437D 03	0.37620D 04	0.10023D 01	0.37630D 04	0.18266D 01
7	7	DUCT B	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
8	8	COMPRESR	0.13000D 01	0.16200D 00	0.51680D 04	0.37070D 04	0.92177D 02	0.37080D 04	0.90596D 00	0.37090D 04	0.18516D 01
9	9	DUCT B	0.48000D 01	0.00000	0.00000	0.00000	0.29900D 04	0.10000D 01	0.18400D 05	0.00000	0.00000
10	10	TURBINE	0.40000D 01	0.79000D 00	0.43996D 00	0.38010D 04	0.68035D 00	0.38020D 04	0.10336D 01	0.10993D 01	0.62508D 00
11	11	DUCT B	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
12	12	TURBINE	0.25000D 01	0.21000D 00	0.37940D 00	0.38030D 04	0.86961D 00	0.38040D 04	0.99700D 00	0.22118D 01	0.00000
13	13	DUCT B	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
14	14	NOZZLE	0.66017D 03	0.10000D 01	0.00000	0.00000	0.99000D 00	0.00000	0.00000	0.00000	0.00000
15	15	SHAFT	0.40000D 04	0.10000D 01							
16	16	SHAFT	0.60000D 04	0.10000D 01							
17	17	DUCT B	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000

CASE IDENTIFICATION SAMPLE CASE FOR PREPWRITE

STATION PROPERTY OUTPUT DATA

FLOW STATION	WEIGHT FLOW	TOTAL PRESSURE	TOTAL TEMPERATURE	FUEL/AIR RATIO	REFERRED FLOW	MACH NUMBER	STATIC PRESSURE	INTERFACE CORRECTED FLOW	FLOW ERROR
	STATP1	STATP2	STATP3	STATP4	STATP5	STATP6	STATP7	STATP8	
1	0.10000D 04	0.14696D 02	0.53867D 03	0.00000	0.10191D 04	0.00000	0.00000	0.14696D 02	0.00000
2	0.10000D 04	0.14696D 02	0.53867D 03	0.00000	0.10191D 04	0.00000	0.00000	0.14696D 02	0.00000
3	0.10000D 04	0.25483D 02	0.64979D 03	0.00000	0.64548D 03	0.00000	0.00000	0.14696D 02	0.00000
4	0.81214D 03	0.25483D 02	0.64979D 03	0.00000	0.52422D 03	0.00000	0.00000	0.14696D 02	0.00000
5	0.81214D 03	0.25483D 02	0.64979D 03	0.00000	0.52422D 03	0.92227D 00	0.14696D 02	0.00000	
6	0.81214D 03	0.25483D 02	0.64979D 03	0.00000	0.52422D 03	0.91304D 00	0.14696D 02	0.00000	
7	0.18786D 03	0.25483D 02	0.64979D 03	0.00000	0.12126D 03	0.00000	0.00000	0.00000	
8	0.18786D 03	0.34988D 02	0.72056D 03	0.00000	0.93005D 02	0.00000	0.00000	0.00000	
9	0.18786D 03	0.34988D 02	0.72056D 03	0.00000	0.93005D 02	0.00000	0.00000	0.00000	
10	0.15743D 03	0.44610D 03	0.15533D 04	0.00000	0.89748D 01	0.00000	0.00000	0.00000	
11	0.16125D 03	0.42468D 03	0.29900D 04	0.24256D-01	0.13397D 02	0.00000	0.00000	0.00000	
12	0.18529D 03	0.98812D 02	0.20965D 04	0.21043D-01	0.55404D 02	0.00000	0.00000	0.00000	
13	0.18529D 03	0.98812D 02	0.20965D 04	0.21043D-01	0.55404D 02	0.00000	0.00000	0.00000	
14	0.19168D 03	0.22886D 02	0.15247D 04	0.20327D-01	0.21104D 03	0.00000	0.00000	0.00000	
15	0.19168D 03	0.22886D 02	0.15247D 04	0.20327D-01	0.21104D 03	0.83092D 00	0.14696D 02	0.00000	
16	0.19168D 03	0.22886D 02	0.15247D 04	0.20327D-01	0.21104D 03	0.82261D 00	0.14696D 02	0.00000	
17	0.30434D 02	0.44610D 03	0.15533D 04	0.00000	0.00000	0.00000	0.00000	0.00000	
19	0.15743D 03	0.44610D 03	0.15533D 04	0.00000	0.89748D 01	0.00000	0.00000	0.00000	

COMPONENT OUTPUT DATA

COMPONENT NO.	TYPE	DATOUT1	DATOUT2	DATOUT3	DATOUT4	DATOUT5	DATOUT6	DATOUT7	DATOUT8	DATOUT9
1	INLET	0.00000	0.00000	0.00000	0.10000D 01	0.10000D 01	0.00000	0.10000D 01	0.10386D 01	0.10000D-04
2	COMPRESR	-0.37786D 05	0.40000D 04	0.00000	0.18000D 01	0.81382D 01	0.10000D 01	0.10227D 04	0.82400D 00	0.17340D 01
3	SPLITTER	0.43230D 01	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
4	DUCT B	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
5	NOZZLE	0.26647D 05	0.10556D 04	0.17340D 01	0.15353D 04	0.15353D 04	0.10000D 01	0.99000D 00	0.18930D 01	0.17340D 01
6	COMPRESR	-0.45423D 04	0.40000D 04	0.00000	0.13000D 01	0.16143D 01	0.10000D 01	0.12437D 03	0.86700D 00	0.13730D 01
7	DUCT B	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
8	COMPRESR	-0.56024D 05	0.60000D 04	0.00000	0.13000D 01	0.32396D 02	0.98500D 00	0.92177D 02	0.85700D 00	0.12750D 02
9	DUCT B	0.00000	0.48000D-01	0.00000	0.24256D-01	0.00000	0.13767D 05	0.00000	0.10000D 01	0.29900D 04
10	TURBINE	0.56024D 05	0.60000D 04	0.10000D 01	0.40000D 01	0.43996D 00	0.56800D 04	0.68035D 00	0.91900D 00	0.42979D 01
11	DUCT B	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
12	TURBINE	0.42328D 05	0.40000D 04	0.10000D 01	0.25000D 01	0.37940D 00	0.52540D 04	0.86961D 00	0.91200D 00	0.43176D 01
13	DUCT B	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
14	NOZZLE	0.87342D 04	0.14660D 04	0.15573D 01	0.64017D 03	0.64017D 03	0.10000D 01	0.99000D 00	0.18616D 01	0.15573D 01
15	SHAFT	0.00000	0.40000D 04	0.40000D 04	0.40000D 04	0.40000D 04	0.00000	0.00000	0.00000	0.00000
16	SHAFT	0.00000	0.60000D 04	0.60000D 04	0.60000D 04	0.00000	0.00000	0.00000	0.00000	0.00000
17	DUCT B	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000

MACH= 0.0000 ALTITUDE= 0. RECOVERY= 1.0000 0 ITERATIONS 2 PASSES

AIRFLOW (LB/SEC)	1000.00	GROSS THRUST	35380.75	FUEL FLOW (LB/HR)	13747.27
NET THRUST	35380.75	TSFC	0.3886	NET THRUST/AIRFLOW	35.3808
TOTAL INLET DRAG	0.00	TOTAL BRAKE SHAFT HP	0.00	BOATTAIL DRAG	0.00
INSTALLED THRUST	35380.75	INSTALLED TSFC	0.3886	SPILLAGE + LIP DRAG	0.00

APPENDIX D

Source Code Listing for PREPWRITE

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IMPLICIT REAL*8 (A-H,O-Z)          0000100
LOGICAL IPLT,PLOT,ISII,ISIO      0000200
DIMENSION ILENG(30),IREST(30),IALL(60),ITYPE(60),ITYPEI(60),I 0000300
1WORD(11),DEFAUL(17,18),DESVAL(17,60),RMIN(17,18),RMAX(17,18), 0000400
2IWIEC(7,60),RMAN(7),RMIX(7)      0000500
DATA IWORD/4HHLT,4HDUCT,4HWINJ,4HCOMP,4HTURB,4HHTEX,4HSPLT,4HMIXR 0000600
1,4HNOZ,4HLOAD,4HSHTF/          0000700
INTEGER AA,BB,CC,DD,EE,FF,GG,HH,OO,PP,QQ,RR,SS,TT,PBURX,DBURX,AUGX 0000800
1,DUCTX,FAHX,FOX,FIX,RSFOX,RSFIX,HPCX,HPTX,HTXX,FMIX,SHAFX 0000900
DATA AA,BB,CC,DD,EE,FF,GG,HH,II,JJ,KK,LL,MM,NN,OO,PP,QQ,RR,SS,TT/ 0001000
1,HA,1HB,1HC,1HD,1HE,1HF,1HG,1HH,1HI,1HJ,1HK,1HL,1HM,1HN,1HO,1HQ 0001100
2,1HR,1HS,1HT/                  0001200
DATA PBURX,DBURX,AUGX,DUCTX/4HPBUR,4HDBUR,4HAUG,4HDUCT/ 0001300
DATA FMIX,MIX,HOZX,SHAFX/4HFMIX,4HNOZ,4HSHAF/ 0001400
DATA FANX,FOX,FIX,RSFOX,RSFIX,LPCX,HPCX/4HFAH,4HFO,4HFI,4HRSF 0001500
10,4HRSFY,4HLPCC,4HHPCC,HPTX,LPTX,HTXX/4HHPT,4HLPT,4HHTEX/ 0001600
DATA DEFAUL/100.,.015,15X0,150.,.015,150.,.300.,.015,15X0,.0,4,1., 0001700
10,-1,13X0,..55,1,7,..45,1,5,4,3,..45,0,0,1,0,2,1,2X0,1,8,..055,..55 0001800
2,055,..55,1,7,..45,1,5,4,3,..45,0,0,1,0,2,1,2X0,1,8,..055,..55 0001900
3,1,7,..45,1,5,4,3,..45,0,0,1,0,2,1,2X0,1,8,..055,..55,1,7,..45 0002000
4,1,5,4,3,..45,0,0,1,0,2,1,2X0,1,8,..055,..55,1,7,..45,1,5,4,..45,0 0002100
53,..45,0,0,1,0,2,1,2X0,1,8,..055,..50,1,5,..40,1,5,4,3,..45,0 0002200
6,..0,1,0,2,1,2X0,1,2,120,40,1,4,..70,1,5,3,1,5,..30,0,0,1, 0002300
7,0,2,1,2X0,1,2,120,3,28,1,5,1,5,1,5,..45,125000,2,1,2X0,..1, 0002400
80,155,3X0,1,0,0,155,..45,28,1,5,2,4,..55,125000,2,1,2X0,..0,19 0002500
95,3X0,1,0,0,195,3,4X0,..45,11X0,..50000,..5,5,14X0,1,8,..15X0,..1 0002600
6,..16X0,..50000,..286,15X0/ 0002700
DATA RMIN/100.,.01,15X0,150.,.01,15X0,.200.,.01,15X0,..4,16X0,..5 0002800
1,1,5,..4,1,3,2,..45,0,0,1,0,0,1,4X0,..5,1,5,..4,1,3,2,..45 0002900
2,0,..0,1,0,0,8,1,4X0,..5,1,5,..4,1,3,2,..45,0,0,1,0,0,1,4 0003000
3X0,..5,1,5,..4,1,3,2,..45,0,0,1,0,0,1,4X0,..5,1,5,..4,1,3,.. 0003100
42,..45,0,0,1,0,0,1,4X0,..45,1,5,..4,1,3,2,..45,0,0,1,0,.. 0003200
50,..1,4X0,..4,1,4,..6,1,2,1,..20,0,0,1,0,0,1,4X0,..3,2,1,.. 0003300
61,..1,..45,100000,0,1,..8X0,..4,2,1,2,4,..55,100000,0,1,..8X0,.. 0003400
7,17X0,0,..3,3,14X0,1,7,15X0,1,1648,40000,..28,15X0/ 0003500
DATA RMAX/150.,..02,15X0,..200.,..02,15X0,..300.,..02,15X0,..5,16X0,..6 0003600
1,1,8,..5,1,5,5,3,..55,0,8,1,0,0,1,4X0,..6,1,8,..5,1,5,5,3,..1 0003700
255,0,0,1,0,0,1,4X0,..6,1,8,..5,1,5,5,3,..55,0,0,1,0,0,0,1 0003800
3,..4X0,..6,1,8,..5,1,5,5,3,..55,0,0,1,0,0,1,4X0,..6,1,8,..5,1, 0003900
45,5,3,..55,0,0,1,0,0,1,4X0,..6,1,8,..5,1,5,5,3,..55,0,0,1 0004000
5,0,0,0,1,4X0,..5,1,7,..8,1,5,5,2,..30,0,0,1,0,0,1,4X0,..4, 0004100
6,3,1,5,2,2,..5,150000,0,1,..8X0,..5,3,1,5,3,6,..6,150000,0,..1 0004200
7,..8X0,..17X0,0,..5,5,14X0,2,9,15X0,2,2,16X0,..50000,..31,..85,14 0004300
8X0,/ 0004400
DATA RMAN,RMIX/14X0,/,IPLT,PLOT,ISII,ISIO/.TRUE.,.TRUE.,.FALSE.,.F 0004500
1ALSE./,IHT,IOUTCD/2,2/ 0004600
WRITE (35,970) 0004700
READ (30,980) IPLT,PLOT,ISII,ISIO 0004800
WRITE (6,990) IPLT,PLOT,ISII,ISIO 0004900
WRITE (35,1000) 0005000
READ (30,1010) IWT,IOUTCD 0005100
WRITE (6,1020) IWT,IOUTCD 0005200
WRITE (35,1210) 0005300
READ (30,1120) IDISK 0005400
IF (IDISK.EQ.88) DISKNI=0. 0005500
IF (IDISK.EQ.66) DISKNI=1. 0005600
WRITE (6,1220) DISKNI 0005700
DO 10 I=1,30 0005800
IUMEC(1,I)=0 0005900
IUMEC(1,I+30)=0 0006000

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10  IREST(I)=0          0006100
    ILENG(I)=0          0006200
    WRITE (35,1030)
    READ (30,1040) ILENG
    WRITE (35,1050)
    READ (30,1040) IREST
    DO 20 I=1,30
    IF (ILENG(I).EQ.0) GO TO 30
20  ILMAX=I          0006400
    WRITE (6,1060) (ILENG(I),I=1,ILMAX)
    DO 40 I=1,30
    IF (IREST(I).EQ.0) GO TO 50
40  IRMAX=I          0006500
    IBEG=ILMAX+1
    IEHD=ILMAX+IRMAX
    DO 60 I=1,60
60  IALL(I)=0          0006600
    DO 70 I=1,ILMAX
70  IALL(I)=ILENG(I)
    DO 80 I=IBEG,IEHD
    IRST=I-IBEG+1
80  IALL(I)=IREST(IRST)
C   IALL CONTAINS ALL THE COMPONENT NUMBERS FOR WHICH WE NEED INPUTS
C   HOW IDENTIFY COMPONENT TYPE PER NNEP GENERIC TYPE
    WRITE (35,1070)
    DO 90 I=1,IEHD
    WRITE (35,1080) IALL(I)
    READ (30,1090) ITYPE(I)
90  CONTINUE
    DO 110 I=1,IEHD
    DO 100 J=1,11
    IF (ITYPE(I).NE.IWORD(J)) GO TO 100
    ITYPEI(I)=J
    GO TO 110
100 CONTINUE
110 CONTINUE
C   ITYPEI HAS COMPONENT TYPE BY NUMERICAL TYPE
    WRITE (35,1100)
    JCX=0          0008400
    JCXX=IALL(JCX)
120 JCXX=JCX+1          0008500
    IF (JCX.GT.IEHD) GO TO 570
    IGO=ITYPEI(JCX)
    GO TO (130,140,240,250,330,390,430,440,480,560,520), IGO
C   IHLET YOU DUMMY
130 GO TO 120          0008600
C   DUCT
140 WRITE (35,1110) IALL(JCX)
    READ (30,1120) IJK
    IF (IJK.EQ.AA) IWMEC(1,JCXX)=PBURX
    IF (IJK.EQ.AA) KT=1
    IF (IJK.EQ.BB) IWMEC(1,JCXX)=DBURX
    IF (IJK.EQ.BB) KT=2
    IF (IJK.EQ.CC) IWMEC(1,JCXX)=AUGX
    IF (IJK.EQ.CC) KT=3
    IF (IJK.EQ.DD) IWMEC(1,JCXX)=DUCTX
    IF (IJK.EQ.DD) KT=4
    IF (KT.EQ.4) GO TO 150
    WRITE (35,1130)
    READ (30,1120) IJK

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IWMEC(2,JCXX)=0          0012100
IF (IJK.EQ.AA) IWMEC(2,JCXX)=1 0012200
GO TO 160                0012300
150  WRITE (35,1140)          0012400
    READ (30,1120) IJK        0012500
    IF (IJK.EQ.AA) IWMEC(2,JCXX)=1 0012600
    IF (IJK.EQ.BB) IWMEC(2,JCXX)=2 0012700
    IF (IJK.EQ.CC) IWMEC(2,JCXX)=3 0012800
    IF (IJK.EQ.DD) IWMEC(2,JCXX)=4 0012900
    IF (IJK.EQ.EE) IWMEC(2,JCXX)=5 0013000
C NOW FILL IN THE DEFAULT VALUES FOR THIS COMPONENT 0013100
160  DO 170 I=1,17          0013200
170  DESVAL(I,JCXX)=DEFAUL(I,KT) 0013300
    KX=JCXX                  0013400
180  IF (KT.GT.1) GO TO 190    0013500
    WRITE (35,1150) AA,DESVAL(1,KX),RMIN(1,KT),RMAX(1,KT),BB,DESVAL(2, 0013600
1KX),RMIN(2,KT),RMAX(2,KT),CC,DESVAL(3,KX),RMIN(3,KT),RMAX(3,KT),DD 0013700
2,DESVAL(4,KX),RMIN(4,KT),RMAX(4,KT),EE,DESVAL(5,KX),RMIN(5,KT),RMA 0013800
3X(5,KT)                  0013900
    GO TO 220                0014000
190  IF (KT.GT.2) GO TO 200    0014100
    WRITE (35,1160) AA,DESVAL(1,KX),RMIN(1,KT),RMAX(1,KT),BB,DESVAL(2, 0014200
1KX),RMIN(2,KT),RMAX(2,KT),CC,DESVAL(3,KX),RMIN(3,KT),RMAX(3,KT),DD 0014300
2,DESVAL(4,KX),RMIN(4,KT),RMAX(4,KT),EE,DESVAL(5,KX),RMIN(5,KT),RMA 0014400
3X(5,KT)                  0014500
    GO TO 220                0014600
200  IF (KT.GT.3) GO TO 210    0014700
    WRITE (35,1170) AA,DESVAL(1,KX),RMIN(1,KT),RMAX(1,KT),BB,DESVAL(2, 0014800
1KX),RMIN(2,KT),RMAX(2,KT),CC,DESVAL(3,KX),RMIN(3,KT),RMAX(3,KT),DD 0014900
2,DESVAL(4,KX),RMIN(4,KT),RMAX(4,KT),EE,DESVAL(5,KX),RMIN(5,KT),RMA 0015000
3X(5,KT)                  0015100
    GO TO 220                0015200
210  CONTINUE                 0015300
    WRITE (35,1200) AA,DESVAL(1,KX),RMIN(1,KT),RMAX(1,KT),BB,DESVAL(2, 0015400
1KX),RMIN(2,KT),RMAX(2,KT),CC,DESVAL(3,KX),RMIN(3,KT),RMAX(3,KT),DD 0015500
2,DESVAL(4,KX),RMIN(4,KT),RMAX(4,KT)                  0015600
220  WRITE (35,960) QQ,RR,TT  0015700
230  CALL REDOEM (IND,DESVAL,JCXX) 0015800
    IF (IND.EQ.QQ.AND.IND.NE.TT.AND.IND.NE.RR) GO TO 230 0015900
    IF (IND.EQ.QQ) GO TO 570 0016000
    IF (IND.EQ.TT) GO TO 120 0016100
    IF (IND.EQ.RR) GO TO 180 0016200
C NO WEIGHT OR DIMENSIONS FOR WATER INJECTORS 0016300
240  CONTINUE                 0016400
    GO TO 120                0016500
250  WRITE (35,950) IALL(JCX)  0016600
    READ (30,1120) IJK        0016700
    IF (IJK.EQ.AA) IWMEC(1,JCXX)=FANX 0016800
    IF (IJK.EQ.AA) KT=5 0016900
    IF (IJK.EQ.BB) IWMEC(1,JCXX)=FOX 0017000
    IF (IJK.EQ.BB) KT=6 0017100
    IF (IJK.EQ.CC) IWMEC(1,JCXX)=FIX 0017200
    IF (IJK.EQ.CC) KT=7 0017300
    IF (IJK.EQ.DD) IWMEC(1,JCXX)=RSFOX 0017400
    IF (IJK.EQ.DD) KT=8 0017500
    IF (IJK.EQ.EE) IWMEC(1,JCXX)=RSFIX 0017600
    IF (IJK.EQ.EE) KT=9 0017700
    IF (IJK.EQ.FF) IWMEC(1,JCXX)=LPCX 0017800
    IF (IJK.EQ.FF) KT=10 0017900
    IF (IJK.EQ.GG) IWMEC(1,JCXX)=IIPCX 0018000

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IF (IJK.EQ.GG) KT=11          0018100
WRITE (35,850)                0018200
READ (30,1120) IJK           0018300
IF (IJK.EQ.AA) IWMEC(2,JCXX)=1 0018400
IF (IJK.EQ.BB) IWMEC(2,JCXX)=0 0018500
IF (IJK.EQ.CC) IWMEC(2,JCXX)=2 0018600
WRITE (35,860)                0018700
READ (30,1120) IJK           0018800
IF (IJK.EQ.AA) IWMEC(3,JCXX)=0 0018900
IF (IJK.EQ.BB) IWMEC(3,JCXX)=1 0019000
IF (IJK.EQ.CC) IWMEC(3,JCXX)=2 0019100
IF (IJK.EQ.DD) IWMEC(3,JCXX)=4 0019200
WRITE (35,870)                0019300
READ (30,1120) IJK           0019400
IF (IJK.EQ.AA) IWMEC(4,JCXX)=0 0019500
IF (IJK.EQ.BB) IWMEC(4,JCXX)=1 0019600
IF (IJK.EQ.CC) IWMEC(4,JCXX)=2 0019700
IF (IJK.EQ.DD) IWMEC(4,JCXX)=4 0019800
IF (KT.LT.6.OR.KT.GT.9) GO TO 260 0019906
WRITE (35,880)                0020000
READ (30,890) IWMEC(5,JCXX) 0020100
260 WRITE (35,900) IWMEC(6,JCXX) 0020200
READ (30,890) IWMEC(6,JCXX) 0020300
WRITE (35,910)                0020400
READ (30,890) IWMEC(7,JCXX) 0020500
C HOW FILL IN THE DEFAULT VALUES FOR THIS COMPONENT 0020600
DO 270 I=1,17                0020700
270 DESVAL(1,JCXX)=DEFAUL(I,KT) 0020800
IF (INMEC(2,JCXX).NE.2) GO TO 290 0020900
DESVAL(4,JCXX)=DESVAL(10,JCXX) 0021000
DESVAL(5,JCXX)=DESVAL(7,JCXX) 0021100
DESVAL(6,JCXX)=1.               0021200
DO 280 IIJ=7,17               0021300
280 DESVAL(IIJ,JCXX)=0.          0021400
GO TO 310                   0021500
C HOW WE UPDATE COMPRESSOR DESIGN VALUES 0021600
290 WRITE (35,930) AA,DESVAL(1,JCXX),RMIN(1,KT),RMAX(1,KT),BB,DESVAL(2 0021700
1,JCXX),RMIN(2,KT),RMAX(2,KT),CC,DESVAL(3,JCXX),RMIN(3,KT),RMAX(3,KT) 0021800
2T),DD,DESVAL(4,JCXX),RMIN(4,KT),RMAX(4,KT),EE,DESVAL(5,JCXX),RMIN( 0021900
35,KT),RMAX(5,KT),FF,DESVAL(6,JCXX),RMIN(6,KT),RMAX(6,KT),GG,DESVAL 0022000
(7,JCXX),RMIN(7,KT),RMAX(7,KT),HH,DESVAL(8,JCXX),RMIN(8,KT),RMAX(8 0022100
5,KT),II,DESVAL(9,JCXX),RMIN(9,KT),RMAX(9,KT),JJ,DESVAL(10,JCXX),RN 0022200
6IH(10,KT),RMAX(10,KT),KK,DESVAL(11,JCXX),RMIN(11,KT),RMAX(11,KT),L 0022300
7L,DESVAL(12,JCXX),RMIN(12,KT),RMAX(12,KT),MM,DESVAL(13,JCXX),RMIN( 0022400
813,KT),RMAX(13,KT),NN,DESVAL(14,JCXX),RMIN(14,KT),RMAX(14,KT),OO, 0022500
9ESVAL(15,JCXX),RMIN(15,KT),RMAX(15,KT),PP,DESVAL(16,JCXX),RMIN(16, 0022600
6KT),RMAX(16,KT),SS,DESVAL(17,JCXX),RMIN(17,KT),RMAX(17,KT) 0022700
WRITE (35,960) QQ,RR,TT          0022800
-300 CALL REDOEM (IND,DESVAL,JCXX) 0022900
IF (IND.NE.QQ.AND.IND.NE.TT.AND.IND.NE.RR) GO TO 300 0023000
IF (IND.EQ.QQ) GO TO 570        0023100
IF (IND.EQ.TT) GO TO 120        0023200
IF (IND.EQ.RR) GO TO 290        0023300
310 WRITE (35,940) AA,DESVAL(1,JCXX),RMAN(1),RMIX(1),BB,DESVAL(2,JCXX) 0023400
1,RMAN(2),RMIX(2),CC,DESVAL(3,JCXX),RMAN(3),RMIX(3),DD,DESVAL(4,JCX 0023500
2X),RMAN(4),RMIX(4),EE,DESVAL(5,JCXX),RMAN(5),RMIX(5),FF,DESVAL(6,J 0023600
3CXX),RMAN(6),RMIX(6),GG,DESVAL(7,JCXX),RMAN(7),RMIX(7) 0023700
WRITE (35,960) QQ,RR,TT          0023800
-320 CALL REDOEM (IND,DESVAL,JCXX) 0023900
IF (IND.NE.QQ.AND.IND.NE.TT.AND.IND.NE.RR) GO TO 320 0024000

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IF (IND.EQ.QQ) GO TO 570          0024100
IF (IND.EQ.TT) GO TO 120          0024200
IF (IND.EQ.RR) GO TO 310          0024300
COME HERE FOR TURBINES          0024400
330  WRITE (35,770) IALL(JCXX)      0024500
READ (30,1120) ITURB             0024600
IF (ITURB.EQ.IH) IWMEC(1,JCXX)=HPTX 0024700
IF (ITURB.EQ.LL) IWMEC(1,JCXX)=LPTX 0024800
WRITE (35,780)                   0024900
READ (30,1120) IJK               0025000
IF (IJK.EQ.AA) IWMEC(2,JCXX)=0    0025100
IF (IJK.EQ.BB) IWMEC(2,JCXX)=1    0025200
WRITE (35,790)                   0025300
READ (30,1040) IWMEC(3,JCXX)      0025400
WRITE (35,800)                   0025500
READ (30,820) IWMEC(4,JCXX)      0025600
WRITE (35,830)                   0025700
READ (30,1040) IWMEC(5,JCXX)      0025800
WRITE (35,840)                   0025900
READ (30,1120) ITURB             0026000
IF (ITURB.EQ.RR) IWMEC(6,JCXX)=2  0026100
IF (ITURB.EQ.AA) IWMEC(6,JCXX)=0  0026200
KT=12                            0026300
IF (IWMEC(1,JCXX).EQ.LPTX) KT=13 0026400
IF (IWMEC(6,JCXX).EQ.2) KT=14    0026500
DO 340 I=1,17                   0026600
340  DESVAL(I,JCXX)=DEFAUL(I,KT)  0026700
IF (KT.EQ.14) GO TO 370          0026800
350  WRITE (35,730) AA,DESVAL(1,JCXX),RMIN(1,KT),RMAX(1,KT),BB,DESVAL(2,JCXX),RMIN(2,KT),RMAX(2,KT),CC,DESVAL(3,JCXX),RMIN(3,KT),RMAX(3,KT),DD,DESVAL(4,JCXX),RMIN(4,KT),RMAX(4,KT),EE,DESVAL(5,JCXX),RMIN(5,KT),RMAX(5,KT),FF,DESVAL(6,JCXX),RMIN(6,KT),RMAX(6,KT),GG,DESVAL(7,JCXX),RMIN(7,KT),RMAX(7,KT),HH,DESVAL(8,JCXX),RMIN(8,KT),RMAX(8,KT),II,DESVAL(9,JCXX),RMIN(9,KT),RMAX(9,KT),JJ,DESVAL(10,JCXX),RMIN(10,KT),RMAX(10,KT),KK,DESVAL(11,JCXX),RMIN(11,KT),RMAX(11,KT),LL,DESVAL(12,JCXX),RMIN(12,KT),RMAX(12,KT),PP,DESVAL(13,JCXX),RMIN(13,KT),RMAX(13,KT),QQ,DESVAL(14,JCXX),RMIN(14,KT),RMAX(14,KT),RR,DESVAL(15,JCXX),RMIN(15,KT),RMAX(15,KT),SS,DESVAL(16,JCXX),RMIN(16,KT),RMAX(16,KT),TT,DESVAL(17,JCXX),RMIN(17,KT),RMAX(17,KT)
1,JCXX),RMIN(2,KT),RMAX(2,KT),CC,DESVAL(3,JCXX),RMIN(3,KT),RMAX(3,KT),DD,DESVAL(4,JCXX),RMIN(4,KT),RMAX(4,KT),EE,DESVAL(5,JCXX),RMIN(5,KT),RMAX(5,KT),FF,DESVAL(6,JCXX),RMIN(6,KT),RMAX(6,KT),GG,DESVAL(7,JCXX),RMIN(7,KT),RMAX(7,KT),HH,DESVAL(8,JCXX),RMIN(8,KT),RMAX(8,KT),II,DESVAL(9,JCXX),RMIN(9,KT),RMAX(9,KT),JJ,DESVAL(10,JCXX),RMIN(10,KT),RMAX(10,KT),KK,DESVAL(11,JCXX),RMIN(11,KT),RMAX(11,KT),LL,DESVAL(12,JCXX),RMIN(12,KT),RMAX(12,KT),PP,DESVAL(13,JCXX),RMIN(13,KT),RMAX(13,KT),QQ,DESVAL(14,JCXX),RMIN(14,KT),RMAX(14,KT),RR,DESVAL(15,JCXX),RMIN(15,KT),RMAX(15,KT),SS,DESVAL(16,JCXX),RMIN(16,KT),RMAX(16,KT),TT,DESVAL(17,JCXX),RMIN(17,KT),RMAX(17,KT)
0026900
0027000
0027100
0027200
0027300
0027400
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0029100
0029200
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0029700
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0029900
0030000

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IF (IDIR.EQ.CC) IWMEC(3,JCXX)=2          0030100
KT=15                                     0030200
DO 400 I=1,17                            0030300
400 DESVAL(I,JCXX)=DEFUAL(I,KT)          0030400
410 WRITE (35,920) AA,DESVAL(1,JCXX),RMIN(1,KT),RMAX(1,KT),BB,DESVAL(2
1,JCXX),RMIN(2,KT),RMAX(2,KT),CC,DESVAL(3,JCXX),RMIN(3,KT),RMAX(3,K
2T),DD,DESVAL(4,JCXX),RMIN(4,KT),RMAX(4,KT) 0030500
2T),DD,DESVAL(4,JCXX),RMIN(4,KT),RMAX(4,KT) 0030600
WRITE (35,960) QQ,RR,TT                  0030700
0030800
420 CALL REDDEM (IND,DESVAL,JCXX)        0030900
IF (IND.NE.QQ.AND.IND.NE.TT.AND.IND.NE.RR) GO TO 420
0031000
IF (IND.EQ.QQ) GO TO 570                0031100
IF (IND.EQ.TT) GO TO 120                0031200
IF (IND.EQ.RR) GO TO 410                0031300
COME HERE IF SPLITTER                  0031400
430 WRITE (35,680) IALL(JCX)             0031500
READ (30,1010) IWMEC(2,JCXX)           0031600
IWMEC(1,JCXX)=IWORD(7)                 0031700
IF (JCXX.HE.ILENG(1)) GO TO 120       0031800
FIRST COMPONENT                         0031900
440 WRITE (35,690)                      0032000
READ (30,700) DESVAL(1,JCXX),DESVAL(2,JCXX)
GO TO 120                                0032100
0032200
COME HERE IF MIXER                      0032300
440 WRITE (35,710) IALL(JCX)             0032400
READ (30,1120) IMIXR                  0032500
IF (IMIXR.EQ.FF) IWMEC(1,JCXX)=FMIX
IF (IMIXR.EQ.CC) IWMEC(1,JCXX)=MIX
0032600
0032700
WRITE (35,720)                         0032800
READ (30,1010) IWMEC(2,JCXX)           0032900
KT=16                                     0033000
DO 450 I=1,17                            0033100
450 DESVAL(I,JCXX)=DEFUAL(I,KT)          0033200
460 WRITE (35,600) AA,DESVAL(1,JCXX),RMIN(1,KT),RMAX(1,KT),BB,DESVAL(2
1,JCXX),RMIN(2,KT),RMAX(2,KT) 0033300
0033400
WRITE (35,960) QQ,RR,TT                  0033500
470 CALL REDDEM (IND,DESVAL,JCXX)        0033600
IF (IND.NE.QQ.AND.IND.NE.TT.AND.IND.NE.RR) GO TO 470
0033700
IF (IND.EQ.QQ) GO TO 570                0033800
IF (IND.EQ.TT) GO TO 120                0033900
IF (IND.EQ.RR) GO TO 460                0034000
COME HERE IF NOZZLE                      0034100
480 WRITE (35,610) IALL(JCX)             0034200
IWMEC(1,JCXX)=NOZX                      0034300
READ (30,1010) IWMEC(2,JCXX)           0034400
WRITE (35,810)                         0034500
READ (30,820) IWMEC(3,JCXX)           0034600
WRITE (35,620)                         0034700
READ (30,1010) IWMEC(4,JCXX)           0034800
KT=17                                     0034900
DO 490 I=1,17                            0035000
490 DESVAL(I,JCXX)=DEFUAL(I,KT)          0035100
500 WRITE (35,630) AA,DESVAL(1,JCXX),RMIN(1,KT),RMAX(1,KT),BB,DESVAL(2
1,JCXX),RMIN(2,KT),RMAX(2,KT) 0035200
0035300
WRITE (35,960) QQ,RR,TT                  0035400
510 CALL REDDEM (IND,DESVAL,JCXX)        0035500
IF (IND.NE.QQ.AND.IND.NE.TT.AND.IND.NE.RR) GO TO 510
0035600
IF (IND.EQ.QQ) GO TO 570                0035700
IF (IND.EQ.TT) GO TO 120                0035800
IF (IND.EQ.RR) GO TO 500                0035900
COME HERE IF SHAFT                      0036000

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520  WRITE (35,640) IALL(JCXX) 0036100
  IWMEC(1,JCXX)=SHAFX
  READ (30,1010) IWMEC(2,JCXX) 0036200
  WRITE (35,650) 0036300
  READ (30,1040) (IWMEC(NN,JCXX),NN=3,6) 0036400
  WRITE (35,660) 0036500
  READ (30,1040) IWMEC(7,JCXX) 0036600
  KT=18 0036700
  DO 530 I=1,17 0036800
  DESVAL(I,JCXX)=DEFAUL(I,KT) 0036900
530  WRITE (35,670) AA,DESVAL(I,JCXX),RMIN(1,KT),RMAX(1,KT),BB,DESVAL(2 0037100
  1,JCXX),RMIN(2,KT),RMAX(2,KT),CC,DESVAL(3,JCXX),RMIN(3,KT),RMAX(3,KT) 0037200
  21,DD,DESVAL(4,JCXX),RMIN(4,KT),RMAX(4,KT),EE,DESVAL(5,JCXX),RMIN( 0037300
  35,KT),RMAX(5,KT),FF,DESVAL(6,JCXX),RMIN(6,KT),RMAX(6,KT),GG,DESVAL 0037400
  47,JCXX),RMIN(7,KT),RMAX(7,KT) 0037500
  WRITE (35,960) QQ,RR,TT 0037600
550  CALL REDOEM (IND,DESVAL,JCXX) 0037700
  IF (IND.NE.QQ.AND.IND.NE.TT.AND.IND.NE.RR) GO TO 550 0037800
  IF (IND.EQ.QQ) GO TO 570 0037900
  IF (IND.EQ.TT) GO TO 120 0038000
  IF (IND.EQ.RR) GO TO 540 0038100
  560  CONTINUE 0038200
  GO TO 120 0038300
570  DO 580 I=1,60 0038400
  IF (IWMEC(1,I).EQ.0) GO TO 580 0038500
  WRITE (6,1180) I,(IWMEC(J,I),J=1,7) 0038600
580  CONTINUE 0038700
  DO 590 I=1,60 0038800
  IF (IWMEC(1,I).EQ.0) GO TO 590 0038900
  WRITE (6,1190) I,(DESVAL(J,I),J=1,17) 0039000
590  CONTINUE 0039100
  WRITE (6,1230) 0039200
  STOP 0039300
C 0039400
600  FORMAT ('0MIXER      /* CODE VALUE DESCRIPTION',31X,'LOW VALU 0039500
  1E-HIGH VALUE/1X,A1,F9.4,' EFF. L/DIAM IF FORCED MIXER ',F20.4,F1 0039600
  20.4/1X,A1,F9.0,' NUMBER OF PASSAGES OR LOBES',F22.0,F10.0) 0039700
610  FORMAT (' COMPONENT ',I3,' IS A NOZZLE, ENTER NOZZLE TYPE/* 1= CO 0039800
  1IVERGENT/* 2= C-D VARIABLE AREA') 0039900
620  FORMAT ('0ENTER THRUST REVERSER TYPE, 0=NONE, 1=FAN, 2=PRIMARY/* 0040000
  1') 0040100
630  FORMAT ('0NOZZLE      /* CODE VALUE DESCRIPTION',31X,'LOW VALU 0040200
  1E-HIGH VALUE/1X,A1,F9.4,' LENGTH TO DIAMETER RATIO ',F20.4,F1 0040300
  20.4/1X,A1,F9.4,' BYPASS RATIO FOR MIXED FLOW',F21.4,F10.4) 0040400
640  FORMAT (' COMPONENT ',I3,' IS A SHAFT, ENTER SHAFT NUMBER FROM INH 0040500
  1ER TO OUTER, I.E. 1 2 3 ETC./* ') 0040600
650  FORMAT ('0ENTER TURBINES CONNECTED TO THIS SHAFT/* 0040700
660  FORMAT ('0ENTER COMPONENT NUMBER OF FIRST COMPRESSOR ON THIS SHAFT 0040800
  1/* ') 0040900
670  FORMAT ('0SHAFT      /* CODE VALUE DESCRIPTION',31X,'LOW VALU 0041000
  1E-HIGH VALUE/1X,A1,F9.2,' SHAFT ALLOWABLE STRESS ',F20.2,F1 0041100
  20.2/1X,A1,F9.2,' SHAFT MATERIAL DENSITY ',F22.2,F10.2/1X,A1,F 0041200
  39.4,' DIAMETER RATIO (INNER/OUTER)',F21.4,F10.4/1X,A1,F9.0,' C 0041300
  40COMPONENT NUMBER ON SHAFT FOR INERTIA',F12.0,F10.0/) 0041400
680  FORMAT (' COMPONENT ',I3,' IS A SPLITTER/* WEIGHT AND LENGTH ARE 0041500
  1IGNORED UNLESS FIRST COMPONENT IN THE ENGINE (AFTER INLET)/* IF I 0041600
  2INNER STREAM IS NOT PRIMARY ENTER A "1" OR ELSE JUST HIT RETURN/* 0041700
  3') 0041800
690  FORMAT (' ENTER MACH NUMBER IN AND HUB TO TIP RATIO IN/* 0041900
  1_____) 0042000

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700  FORMAT (F6.0,F7.0) 0042100
710  FORMAT (' COMPONENT ',I3,' IS A MIXER, ENTER TYPE'/' F- FORCED MIX 0042200
1ER (MECHANICAL - DAISY LOBE TYPE)''/ C- COANNULAR EMERGENCE (HOT F 0042300
20RCED)') 0042400
720  FORMAT (' ENTER PRIMARY INPUT NODE, 0=INNER, 1=OUTER'/' ') 0042500
730  FORMAT ('TURBINE  '' CODE VALUE DESCRIPTION',31X,'LOW VALU 0042600
1E-HIGH VALUE'/'IX,A1,F9.4,' FACE INLET MACH NUMBER ,F20.4,F1 0042700
20.4/IX,A1,F9.5,' LOADING PARAMETER ',F27.5,F10.5/IX,A1,F9.3,' 0042800
3 BLADE SOLIDITY (CORD/SPACING) 'F5.3,F11.3/IX,A1,F9. 0042900
44,' BLADE ASPECT RATIO- 1ST STAGE ',F15.4,F11.4/IX,A1,F9.4,31H 0043000
5 BLADE ASPECT RATIO-LAST STAGE,F19.4,F11.4/IX,A1,F9.4,32H TURBIN 0043100
6E EXIT MACH NUMBER ,F18.4,F11.4/IX,A1,F9.0,31H DISC REFERENC 0043200
7E STRESS ,F19.0,F11.0/IX,A1,F9.0,32H 1=CONST TIP DESH,2=ME 0043300
8AH,3=HUB ,F18.0,F11.0/IX,A1,F9.4,32H MAX. SPEED RATIO RPMMAX/RPMD 0043400
9 ,F18.4,F11.4/IX,A1,F9.4,31H CONTROL RADIUS- 0 IF TRANSFRD,F19.4 0043500
6,F11.4/IX,A1,F9.4,32H BLADE MATERIAL DEHSITY 0=T-SET,F18.4,F11.4/ 0043600
61X,A1,F9.4,32H BLADE VOLUME FACTOR ,F18.4,F11.4/IX,A1,F 0043700
69.4,32H BLADE TAPER RATIO ,F18.4,F11.4/IX,A1,F9.4,32H 0043800
6 STATOR BLADE VOLUME FACTOR ,F18.4,F11.4) 0043900
740  FORMAT ('TURBINE  '' CODE VALUE DESCRIPTION',31X,'LOW VALU 0044000
1E-HIGH VALUE'/'IX,A1,F9.4,' FACE INLET MACH NUMBER ,F20.4,F1 0044100
20.4/IX,A1,F9.4,' EXIT MACH NUMBER ',F26.4,F10.4) 0044200
750  FORMAT (' COMPONENT ',I3,' IS A HEAT EXCHANGER, OPTIONS ARE'/' F- 0044300
1FIXED OR'/' R- ROTARY') 0044400
760  FORMAT (' FLOW DIRECTION IS'/' P- PARALLEL'/' C- COUNTER FLOW') 0044500
770  FORMAT (' COMPONENT ',I3,' IS A TURBIHE, OPTIONS ARE (ENTER CORRE 0044600
1CT LETTER)'/' H- HIGH PRESSURE TURBINE'/' L- LOW PRESSURE TURBINE' 0044700
2)
780  FORMAT (' A- NO FRAME'/' B- TURBINE HAS EXIT FRAME) 0044800
790  FORMAT (' ENTER COMPONENT NUMBER OF COMPRESSOR THAT SETS TURBINE R 0044900
1PM'/' ')
800  FORMAT (' ENTER COMPONENT NUMBER THAT LIMITS TURBINE OUTER RADIUS' 0045000
1/' (+ = OUTLET, - = INLET,0 = FEEDING COMPONENT)') 0045100
810  FORMAT (' ENTER COMPONENT NUMBER THAT LIMITS NOZZLE OUTER RADIUS' 0045200
1' (+ = OUTLET, - = INLET,0 = FEEDING COMPONENT)') 0045300
820  FORMAT (I3) 0045400
830  FORMAT (' YOU MAY ENTER THE NUMBER OF STAGES IF DESIRED, OTHERWISE 0045500
1 ENTER 0'/' ')
840  FORMAT (' R- RADIAL FLOW TURBINE'/' A- AXIAL FLOW TURBINE') 0045600
850  FORMAT ('0INDICATE WHETHER THERE ARE STATORS OR IF THIS IS A CENTR 0045700
1IFUGAL COMPRESSOR'/' A- CALCULATE STATOR WEIGHT'/' B- NO STATOR WE 0045800
2IGHT'/' C- CENTRIFUGAL COMPRESSOR') 0045900
860  FORMAT ('0 INDICATE WHETHER THERE IS A FRONT FRAME IN THE COMPRESS 0046000
1OR'/' A- NO FRAME'/' B- SGHL BEARING FRAME FOR TFS AND TJS WITHOUT 0046100
2 POWER TAKEOFF (PTO)'/' C- SINGLE BEARING FRAME WITH PTO'/' D- TWO 0046200
3 BEARING FRAME WHICH EXTENDS OUTWARD TO THE FAN OUTER CASE AND HOLD 0046300
4S TWO BEARINGS WITH PTO') 0046400
870  FORMAT ('0 INDICATE WHETHER THERE IS A REAR FRAME IN THE COMPRESS 0046500
1OR'/' A- NO FRAME'/' B- SGHL BEARING FRAME FOR TFS AND TJS WITHOUT 0046600
2 POWER TAKEOFF (PTO)'/' C- SINGLE BEARING FRAME WITH PTO'/' D- TWO 0046700
3 BEARING FRAME WHICH EXTENDS OUTWARD TO THE FAN OUTER CASE AND HOLD 0046800
4S TWO BEARINGS WITH PTO') 0046900
880  FORMAT ('0SPLIT FLOW COMPRESSOR - ENTER COMPONENT NUMBER OF OTHER 0047000
1HALF OF COMPRESSOR'/' ')
890  FORMAT (I2) 0047100
900  FORMAT ('0GEAR BOX INDICATOR,0= NO GEAR BOX, N=SHAFT NUMBER FOR GE 0047200
1AR BOX'/' ')
910  FORMAT ('0YOU MAY SPECIFY THE NUMBER OF STAGES-OVERRIDES MAX PR/ST 0047300
1G., OTHERWISE ENTER 0'/' ')
920  FORMAT ('0HEAT EXCHANGER'/' CODE VALUE DESCRIPTION',31X,'LOW VALU 0047400

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1E-HIGH VALUE'/IX,A1,F9.0,' NUMBER OF TUBES IF "FIXED" ',F20.0,F1 0048100
20.0/IX,A1,F9.4,' MACH NUMBER - PRIMARY ',F27.4,F10.4/IX,A1,F9.4,' 0048200
3 MACH NUMBER-IN SECONDARY ',F5.4,F11.4/IX,A1,F9. 0048300
44,' BYPASS RATIO IF "ROTARY" ',F15.4,F11.4) 0048400
930 FORMAT ('0FAH/COMPRESSOR'/' CODE VALUE DESCRIPTION',3IX,'LOW VALU 0048500
1E-HIGH VALUE'/IX,A1,F9.4,' FACE INLET MACH NUMBER ',F20.4,F1 0048600
20.4/IX,A1,F9.5,' MAX 1ST STAGE PRATIO ',F27.5,F10.5/IX,A1,F9.3,' 0048700
3 COMPRESSOR FACE HUB TO TIP RATIO ',F5.3,F11.3/IX,A1,F9. 0048800
44,' BLADE SOLIDITY (CORD/SPACING) ',F15.4,F11.4/IX,A1,F9.4,31H 0048900
5 BLADE ASPECT RATIO-1ST STAGE ,F19.4,F11.4/IX,A1,F9.4,32H BLADE 0049000
6ASPECT RATIO-LAST STAGE ,F18.4,F11.4/IX,A1,F9.4,31H COMPRESSOR EX 0049100
7IT MACH NUMBER ,F19.4,F11.4/IX,A1,F9.4,32H MAX. COMP. INLET T- 0 0049200
8=CALC'D ,F18.4,F11.4/IX,A1,F9.4,32H MAX. COMP. EXIT T- 0=CALC'D 0049300
9 ,F18.4,F11.4/IX,A1,F9.4,31H MAX. SPEED RATIO RPMMAX/RPMD ,F19.4 0049400
, ,F11.4/IX,A1,F9.4,32H BLADE MATERIAL DENSTY 0=T-SET,F18.4,F11.4/ 0049500
61X,A1,F9.0,32H 1=CONST.HUB,2=MEAN,3=TIP DESH.,F18.0,F11.0/IX,A1,F 0049600
69.4,32H RPM SCALER TO MATCH KNOWN RPM ,F18.4,F11.4/IX,A1,F9.4,32H 0049700
8 TEMP FOR MATERIAL CHANGE ,F18.4,F11.4/IX,A1,F9.4,31H WEIGH 0049800
6T SCALER 0=AS CALCULATED,F19.4,F11.4/IX,A1,F9.4,32H STATOR BLADE 0049900
6TAPER RATIO ,F18.4,F11.4/IX,A1,F9.4,32H BLADE VOLUME RATIO 0050000
8 ,F18.4,F11.4) 0050100
940 FORMAT ('OCENTRIFUGAL H'/' CODE VALUE DESCRIPTION',3IX,'LOW VALU 0050200
1E-HIGH VALUE'/IX,A1,F9.4,' FACE INLET MACH NUMBER ',F22.4,F1 0050300
21.4/IX,A1,F9.5,' MAX 1ST STAGE PRATIO ',F29.5,F10.5/IX,A1,F9.3,' 0050400
3 COMPRESSOR FACE HUB TO TIP RATIO ',F6.3,F11.3/IX,A1,F9 0050500
4.4,' RPM RATIO - RPMMAX/RPMD ',F15.4,F11.4/IX,A1,F9.4,31 0050600
5H COMPRESSOR EXIT MACH NUMBER ,F19.4,F11.4/IX,A1,F9.4,32H GEAR 0050700
6 RATIO OF POWER SHAFT ,F18.4,F11.4/IX,A1,F9.4,31H HORSEPOWER 0 0050800
7F POWER SHAFT ,F19.4,F11.4) 0050900
950 FORMAT (' COMPONENT ',13,' IS A COMPRESSOR, OPTIONS ARE (ENTER CO 0051000
1RECT LETTER)'' A- TYPICAL FAN'' B- OUTER PORTION OF NON-ROTATIN 0051100
2G SPLITTER FAN'' C- INNER PORTION OF NON-ROTATING SPLITTER FAN'' 0051200
3 D- OUTER PORTION OF ROTATING SPLITTER FAN'' E- INNER PORTION OF 0051300
4ROTATING SPLITTER FAN'' F- LOW PRESSURE COMPRESSOR'' G- HIGH PRE 0051400
5SURE COMPRESSOR') 0051500
960 FORMAT (1H0,'ENTER LETTER FOLLOWED BY VALUE (INCLUDING DECIMAL POI 0051600
1HT) OR ENTER'/IX,A1,11X,'QUIT PROCESSING ENTIRE ENGINE'/IX,A1,11X, 0051700
2'REVIEW UPDATED VALUES'/IX,A1,11X,'TERMINATE - GO ON TO NEXT COMPO 0051800
3HEHT') 0051900
970 FORMAT ('0ENTER VALUES FOR IPLT(PRINTER PLOT), PLOT(GRAPHICS PLOT) 0052000
1, ISII(SI INPUT), ISIO(SI OUTPUT) - T OR F', ' _ _ _ ') 0052100
980 FORMAT (4L2) 0052200
990 FORMAT (' 8W IPLT=',L2,'PLOT=',L2,'ISII=',L2,'ISIO=',L2,' ') 0052300
1000 FORMAT ('0ENTER VALUES FOR IWT (2=NO AIRFLOW SCALING,4=WITH AIRFLO 0052400
1W SCALING) AND/,' IOUTCD (0=WT,L, & MAX R,1=ADD COMPONENT WTS,2=A 0052500
2DD STAGE BY STAGE OUTPUT', ' _ _ ') 0052600
1010 FORMAT (1I1,I2) 0052700
1020 FORMAT (' IWT=',I1,'IOUTCD',I1,IH,) 0052800
1030 FORMAT ('0ENTER COMPONENT NUMBERS OF ALL COMPONENTS (EXCLUSIVE OF 0052900
1INLETS AND WINJS) THAT CONTRIBUTE TO MAX LENGTH OF ENGINE (RT. ADJ 0053000
2.)', ' _ _ _ _ _ ') 0053100
3 ' _ _ _ _ _ ') 0053200
1040 FORMAT (12,29I3) 0053300
1050 FORMAT ('0ENTER COMPONENT NUMBERS OF ALL REMAINING COMPONENTS (EXC 0053400
1LUSIVE OF INLETS,WINJS,LOADS,CHTLS,OPTVS, & LIMVS)'' , ' _ _ _ _ _ 0053500
2 ' _ _ _ _ _ ') 0053600
3 ' _ _ _ _ _ ') 0053700
1060 FORMAT (' ILENG(1)=',30(I2,1H,)) 0053800
1070 FORMAT ('0YOU WILL NOW BE PROMPTED FOR THE GENERIC (NHEP) TYPE FOR 0053900
1 EACH COMPONENT (COMP,TURB, ETC.)') 0054000

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1080 FORMAT ('0COMPONENT',I3,' IS A (A4)"/, ' ____') 0054100
1090 FORMAT (A4) 0054200
1100 FORMAT ('0FOR EACH NHEP COMPONENT YOU WILL BE PROMPTED FOR THE WAT 0054300
1E SUBCOMPONENT') 0054400
1110 FORMAT (' COMPONENT ',I3,' IS A DUCT, OPTIONS ARE (ENTER CORRECT L 0054500
1LETTER)"/, ' A- PRIMARY BURNER"/, ' B- DUCT BURGER"/, ' C- AUGMENTOR"/
2,' D- DUCT"/, ' ENTER LETTER"/, ' _') 0054600
1120 FORMAT (A1) 0054700
1130 FORMAT (' A- BURNER HAS A FRAME"/, ' B- NO FRAME"/, ' ') 0054800
1140 FORMAT (' TYPE OF DUCT"/, ' A- DUMMY - NO WT OR LENGTH"/, ' B- INPUT 0055000
1 LENGTH ( WILL SPECIFY L/D OF DUCT )"/, ' C- LENGTH FROM CONNECTION 0055100
2 BETWEEN SPLITTER AND MIXER"/, ' D- CROSSOVER DUCT FOR", ' CENTRIFUGA 0055200
3L COMPRESSORS"/, ' E- DIFFUSER FOR CENTRIFUGAL COMPRESSORS"/, ' ') 0055300
1150 FORMAT ('0PRIMARY BURNER"/, ' CODE VALUE DESCRIPTION',32X,'LOW VALU 0055400
1E-HIGH VALUE'/IX,A1,F9.2,' BURNER THRU-FLOW', ' VELOCITY',F25.2,F1 0055500
20.2/IX,A1,F9.5,' AIRFLOW RESIDENCY TIME',F29.5,F10.5/IX,A1,F9.3,' 0055600
3 MEAN DIAMETER - IF 0 MATCH" UPSTREAM CMPT",F6.0,F11.0/IX,A1,F9 0055700
4.0,' CMPT NUMBER FOR MATCHING DIAMETER',F15.0,F11.0/IX,A1,F9.0,' 0055800
5 NUMBER OF CANS FOR CAN BURNER',F19.0,F11.0) 0055900
1160 FORMAT ('0DUCT BURNER"/, ' CODE VALUE DESCRIPTION',35X,'LOW VALUE-H 0056000
1HIGH VALUE'/IX,A1,F9.2,' BURNER THRU-FLOW', ' VELOCITY',F25.2,F10.2 0056100
2/IX,A1,F9.5,' AIRFLOW RESIDENCY TIME',F29.5,F10.5/IX,A1,F9.3,' M 0056200
3EAN DIAMETER - IF 0 MATCH" UPSTREAM CMPT",F6.0,F11.0/IX,A1,F9.0,' 0056300
4 CMPT NUMBER FOR MATCHING DIAMETER',F15.0,F11.0/IX,A1,F9.0,' NU 0056400
5MBER OF CANS FOR CAN BURNER',F19.0,F11.0) 0056500
1170 FORMAT ('0AUGMENTOR"/, ' CODE VALUE DESCRIPTION',37X,'LOW VALUE-HIG 0056600
1H VALUE'/IX,A1,F9.2,' BURNER THRU-FLOW', ' VELOCITY',F25.2,F10.2/1 0056700
2X,A1,F9.5,' AIRFLOW RESIDENCY TIME',F29.5,F10.5/IX,A1,F9.3,' MEA 0056800
3H DIAMETER - IF 0 MATCH" UPSTREAM CMPT",F6.0,F11.0/IX,A1,F9.0,' 0056900
4 CMPT NUMBER FOR MATCHING DIAMETER',F15.0,F11.0/IX,A1,F9.0,' NUMB 0057000
5ER OF CANS FOR CAN BURNER',F19.0,F11.0) 0057100
1180 FORMAT (1X,'IUMEC(1,',I2,')=4H',A4,'.',6(I3,1H,)) 0057200
1190 FORMAT (1X,'DESVAL(1,',I2,')=',7(G10.4,1H,)/2X,10(G10.4,1H,)) 0057300
1200 FORMAT ('0DUCT "/, ' CODE VALUE DESCRIPTION',37X,'LOW VALUE-HIG 0057400
1H VALUE'/IX,A1,F9.2,' DUCT MACH NUMBER  ',F25.2,F10.2/1 0057500
2X,A1,F9.5,' LEN/HT IF MODE B  ',F33.5,F10.5/IX,A1,F9.3,' DUCT 0057600
3 MEAN DIAM. IF =0 CALC. PER NODE BELOW ',F9.0,F11.0/IX,A1,F9.0,' I 0057700
4F NODE,0=MEAN D SPECIFIED,-1=COHN. COMP.",F11.0,F11.0) 0057800
1210 FORMAT ('0DO YOU WANT BOEING OR GARRETT METHOD FOR DISK WEIGHT CAL 0057900
1CULATIONS?"/, ' ENTER B FOR BOEING OR G FOR GARRETT"/, ' ') 0058000
1220 FORMAT (' DISKWI=',F2.0,',ACCS=0.1,ISCALE(1)=2,3,SCALE(1)=1.,.8,1 0058100
1.2,ACCARM=0,') 0058200
1230 FORMAT (' &END')
EHD 0058300
SUBROUTINE REDOEM(IX,DESVAL,JCXX) 0058400
IMPLICIT REAL*8 (A-H,O-Z) 0058500
DIMENSION DESVAL(17,60) 0058600
INTEGER AA,BB,CC,DD,EE,FF,GG,HH,OO,PP,QQ,RR,SS,TT,PBURX,DBURX,AUGX 0058800
1,DUCTX 0058900
DATA AA,BB,CC,DD,EE,FF,GG,HH,II,JJ,KK,LL,MM,NN,OO,PP,QQ,RR,SS,TT/1 0059000
1HA,1HB,1HC,1HD,1HE,1HF,1HG,1HH,1HI,1HJ,1HK,1HL,1HM,1HN,1HO,1HP,1HQ 0059100
2,1HR,1HS,1HT/
READ (30,10) IX,VALUE 0059200
10 FORMAT (A1,F20.0) 0059300
IF (IX.EQ.AA) DESVAL( 1,JCXX)=VALUE 0059400
IF (IX.EQ.BB) DESVAL( 2,JCXX)=VALUE 0059500
IF (IX.EQ.CC) DESVAL( 3,JCXX)=VALUE 0059600
IF (IX.EQ.DD) DESVAL( 4,JCXX)=VALUE 0059700
IF (IX.EQ.EE) DESVAL( 5,JCXX)=VALUE 0059800
IF (IX.EQ.FF) DESVAL( 6,JCXX)=VALUE 0059900
IF (IX.EQ.GG) DESVAL( 7,JCXX)=VALUE 0060000

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IF (IX.EQ.GG) DESVAL(7,JCXX)=VALUE	0060100
IF (IX.EQ.HH) DESVAL(8,JCXX)=VALUE	0060200
IF (IX.EQ.II) DESVAL(9,JCXX)=VALUE	0060300
IF (IX.EQ.JJ) DESVAL(10,JCXX)=VALUE	0060400
IF (IX.EQ.KK) DESVAL(11,JCXX)=VALUE	0060500
IF (IX.EQ.LL) DESVAL(12,JCXX)=VALUE	0060600
IF (IX.EQ.MM) DESVAL(13,JCXX)=VALUE	0060700
IF (IX.EQ.NH) DESVAL(14,JCXX)=VALUE	0060800
IF (IX.EQ.OO) DESVAL(15,JCXX)=VALUE	0060900
IF (IX.EQ.PP) DESVAL(16,JCXX)=VALUE	0061000
IF (IX.EQ.SS) DESVAL(17,JCXX)=VALUE	0061100
RETURN	0061200
END	0061300

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2. Fishbach, L. H.; and Caddy, M. J.: NNEP: The Navy/NASA Engine Program. NASA TM X-71857, 1975.
3. Fishbach, L. H.: KONFIG and REKONFIG: Two Interactive Preprocessing Programs to the Navy/NASA Engine Program (NNEP). NASA TM-82636, 1981.
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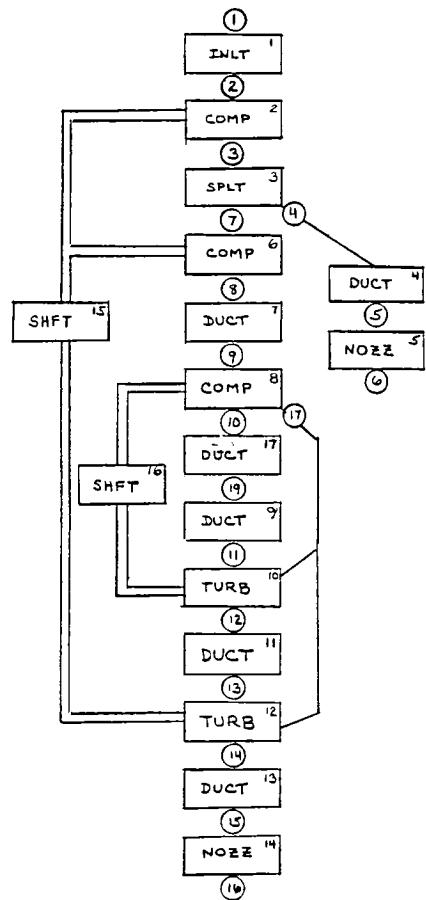


Figure 1. - Thermodynamic schematic for NNEP.

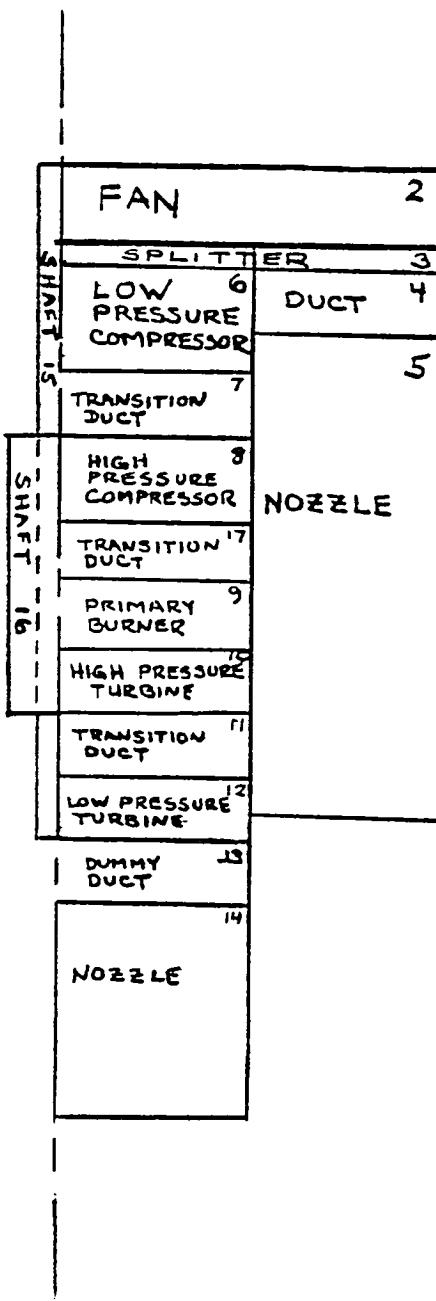


Figure 2. - "Actual" mechanical schematic for WATE by subcomponent type.

COMP#	TYPE	NSTAGE	HT
1	INLT	0	0.
2	FAN	1	1455.
3	SPLT	0	0.
6	LPC	1	314.
7	DUCT	0	12.
8	HPC	14	682.
17	DUCT	0	22.
9	PBUR	0	329.
10	HPT	2	169.
11	DUCT	0	0.
12	LPT	3	1109.
13	DUCT	0	0.
14	NOZ	0	156.
4	DUCT	0	12.
5	NOZ	0	243.

TOTAL ENGINE HT = 4800.

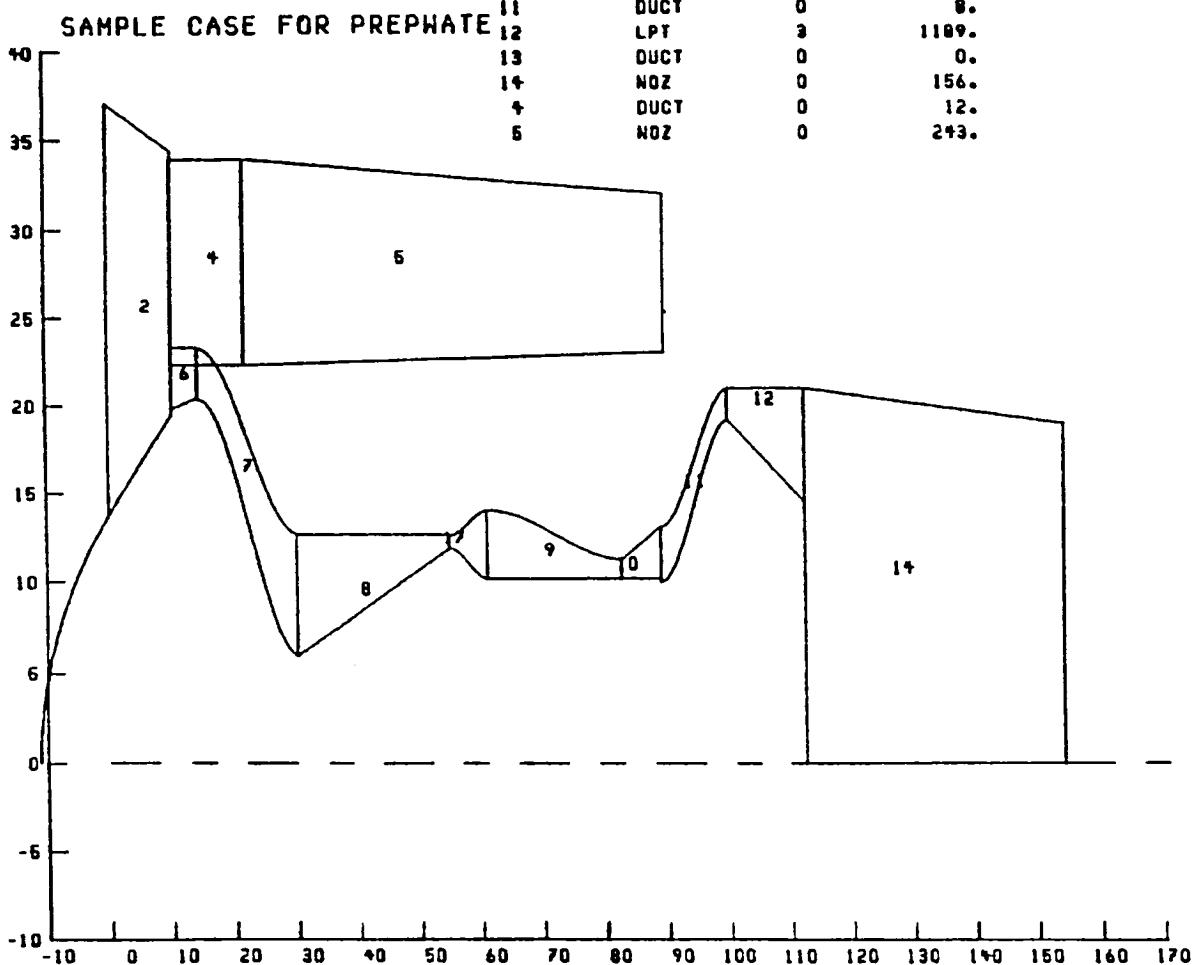


Figure 3. - Graphics plot of engine flowpath.

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7. Author(s) Laurence H. Fishbach		6. Performing Organization Code 505-40-82	
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16. Abstract The Weight Analysis of Turbine Engines (WATE) computer code was developed by Boeing under contract to NASA Lewis (ref. 1). It was designed to function as an adjunct to the Navy/NASA Engine Program (NNEP) (ref. 2). NNEP calculates the design and off-design thrust and sfc performance of User defined engine cycles. The thermodynamic parameters throughout the engine as generated by NNEP are then combined with input parameters defining the component characteristics in WATE to calculate the bare engine weight of this User defined engine. Preprocessor programs for NNEP were previously developed (ref. 3) to simplify the task of creating input datasets. This report describes a similar preprocessor for the WATE code.			
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